



# Investigating Storm-Driven Thermospheric Density Enhancements with Two-Line Element Sets and Orbital Propagation

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## INTRODUCTION

### Thermosphere Dynamics

- Neutral densities increase up to 800% during geomagnetic storms<sup>4</sup>.
- Satellite two-line element sets (TLEs) show increased orbital decay during geomagnetic storms from increased drag<sup>2</sup>.

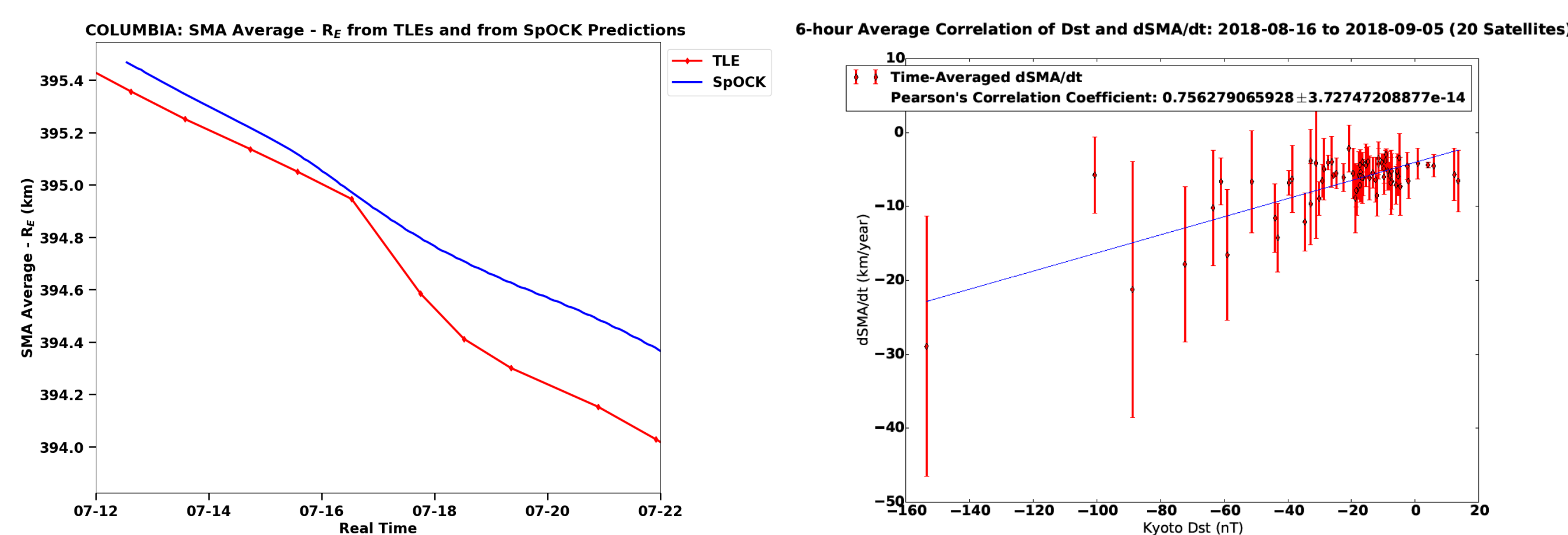


Figure 1: SpOCK inaccurately modeling orbital decay of the Columbia CubeSat.

Figure 2: The strongly positive correlation between rate of deorbit and geomagnetic activity indicated by Dst for 20 identical Flock 2K CubeSats.

### Weakness of Empirical Atmospheric Models

- Models like NRLMSISE-00 (Picone et al. 2002) poorly reproduce the storm-time density increase.
- Poor model performance reduces accuracy of orbital propagators (Figure 1) like UofM's Spacecraft Orbital Characterization Kit (SpOCK)<sup>1</sup>.

### Model Correction

- High-Accuracy Satellite Drag Model (HASDM) used by USAF to correct Jacchia-1970<sup>6</sup>.
- HASDM Dynamics Calibration Atmosphere (DCA) uses Space Surveillance Network data of >75 orbiting spheres to estimate corrections to F10.7 and  $a_p$ <sup>6</sup>.
- Doornbos et al. 2008 used TLEs to estimate corrections but assumed TLE-derived densities were sufficient<sup>3</sup>.
- We propose a method that estimates corrections by minimizing orbit error between SpOCK orbits and TLEs.

## OBJECTIVES

- Develop an algorithm capable of **estimating corrections to empirical model densities** during geomagnetic storms.
- Validate** the corrected densities returned by the algorithm in comparison to *in-situ* densities measured by the SWARM spacecrafts.
- Demonstrate the algorithm's **self-consistency** across a wide variety of modeled spacecraft orbits during different storms.
- Demonstrate the efficacy of using **orbit error minimization** to back out corrected densities from empirical atmospheric models.

## METHODOLOGY

### Multifaceted Optimization Algorithm:

Corrects NRLMSISE-00 model densities:

#### 1. Area Optimization Algorithm (AROPT):

- Loop over preceding quiet time and adjust cross-sectional area until orbit error is minimized.

#### 2. F10.7 Optimization Algorithm (FOPT):

- Obtain the **mean** of the optimized area distribution (assumes NRLMSISE-00 underpredicts effects from storms)
- Repeat the loop, adjusting F10.7 until orbit error is minimized; retrieve the F10.7 correction for each interval.

#### 3. Applying Corrections:

- Linearly interpolate median corrections across all satellites.
- Apply corrections to F10.7 inputs to NRLMSISE-00 along-track the orbits of validation spacecraft.
- Compare the resulting densities to *in-situ* measurements.

### Scenario:

- Time: 2018-08-21 and 2018-08-31
  - Calibration Targets: 10 Flock 2K satellites
  - Validation Satellites: SWARM-A, -B, -C
- Note:** Chosen quartile of the optimized area distribution strongly impacts corrections. Best results when correcting NASA OMNIWeb inputs (static daily indices).



Figure 3: An example of a Flock 2K 3U CubeSat launched by Planet Labs, Inc.

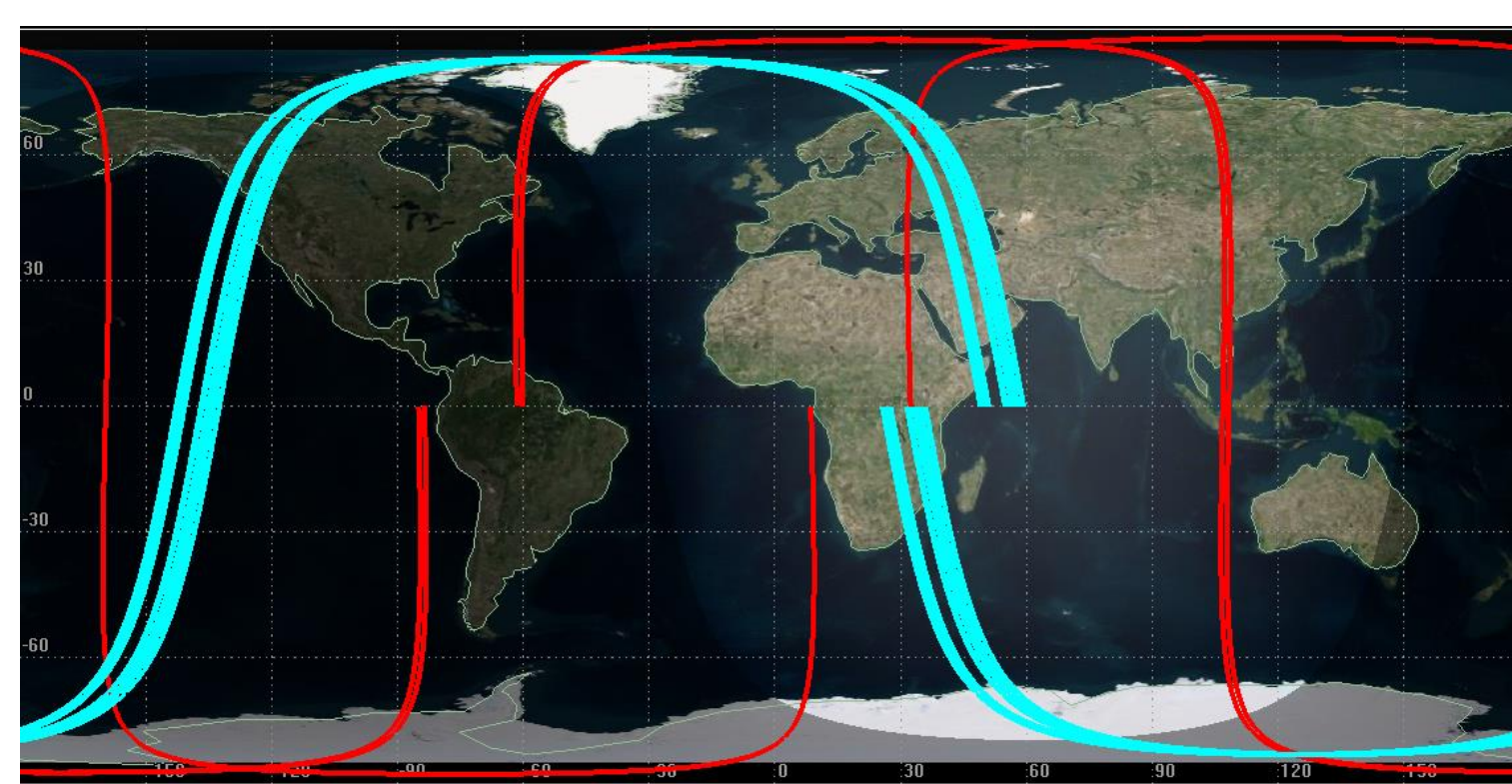


Figure 4: Orbits of the Flock 2K CubeSats (cyan) and of SWARM (red) in the August 2018 storm.

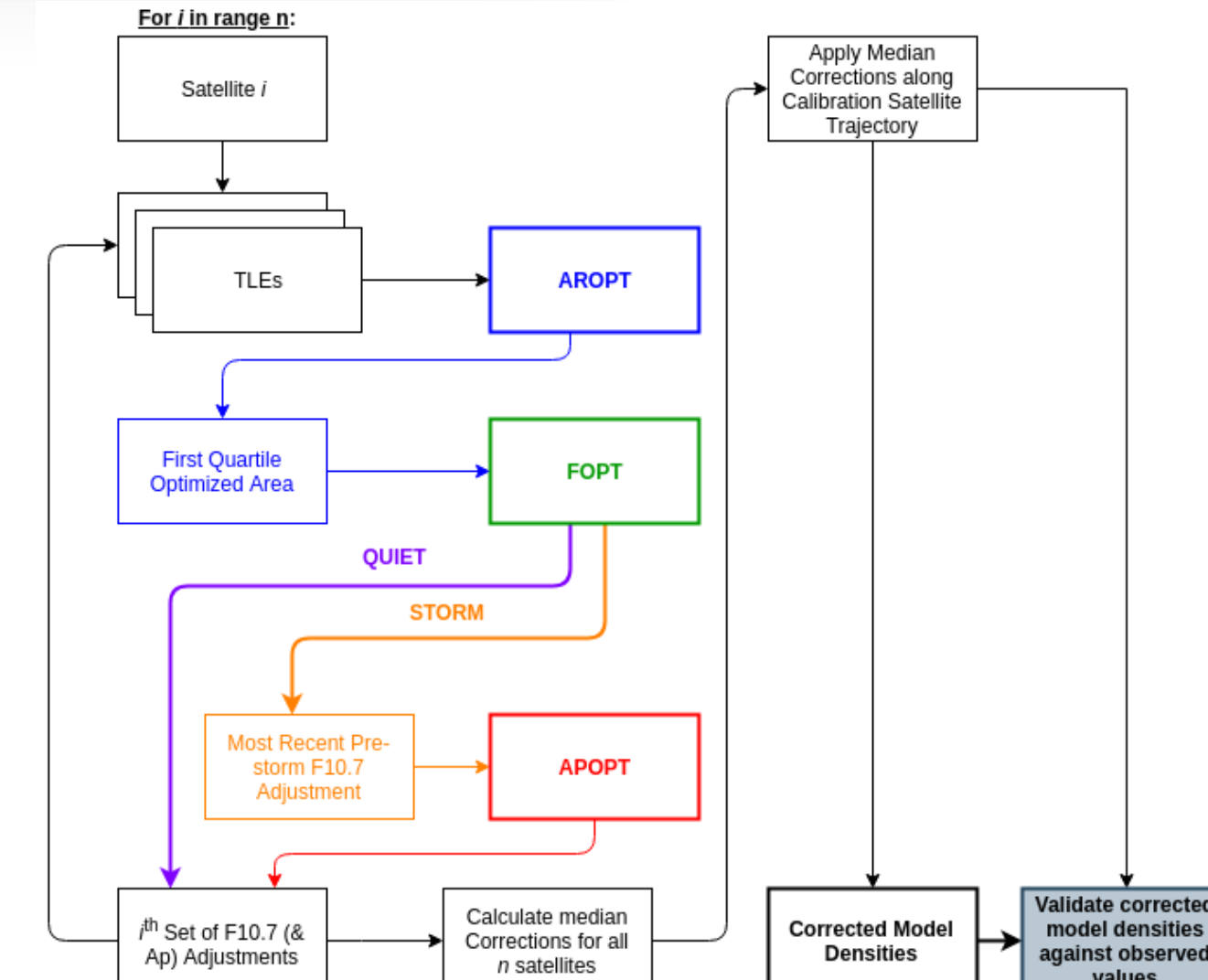


Figure 5: A flowchart of the MOA algorithm. We restricted ourselves to the AROPT and FOPT sub-processes for this initial study.

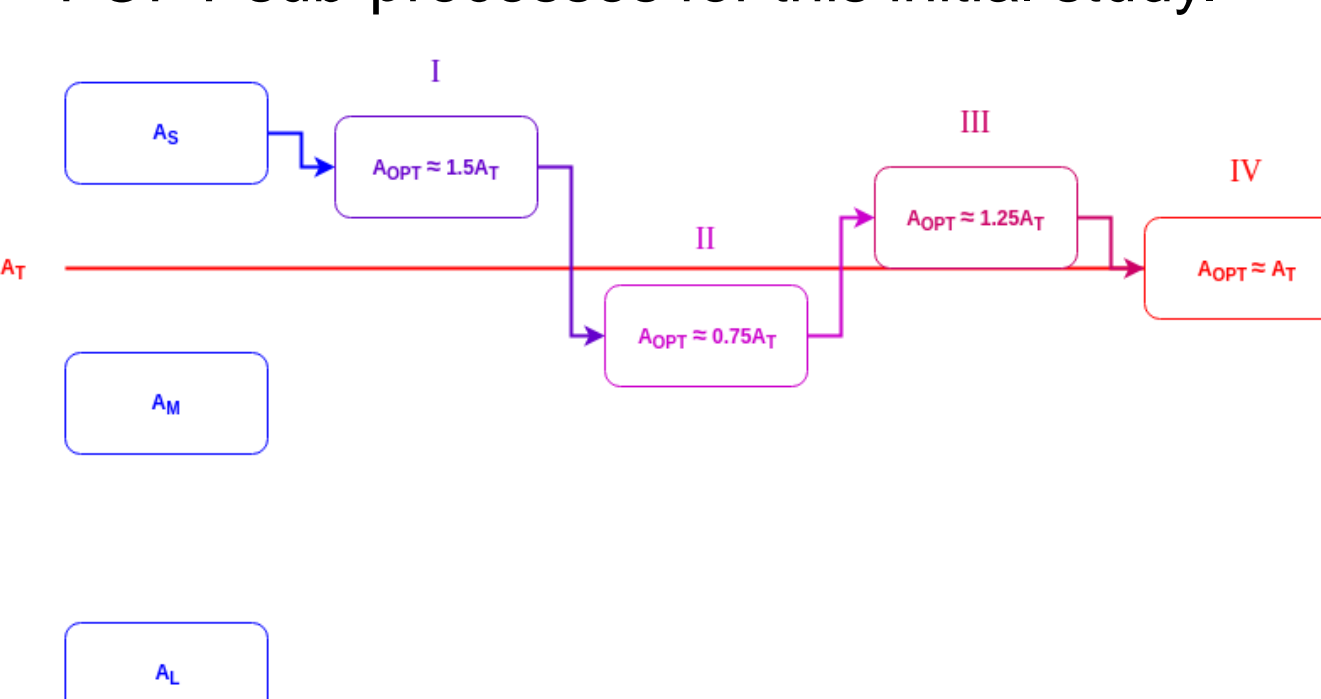


Figure 6: A flowchart of the bracketing processes used by AROPT to find the optimized area.

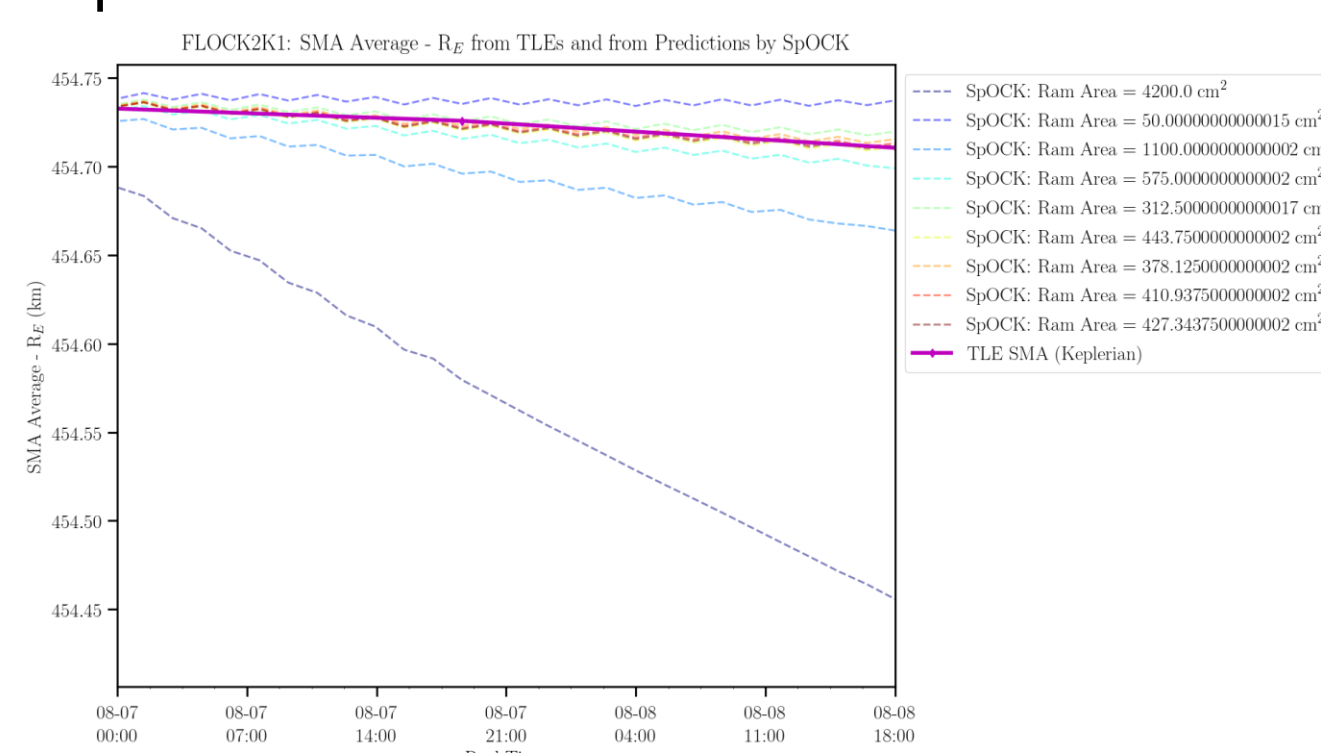


Figure 7: AROPT optimizing the area of the Flock 2K 1 satellite.

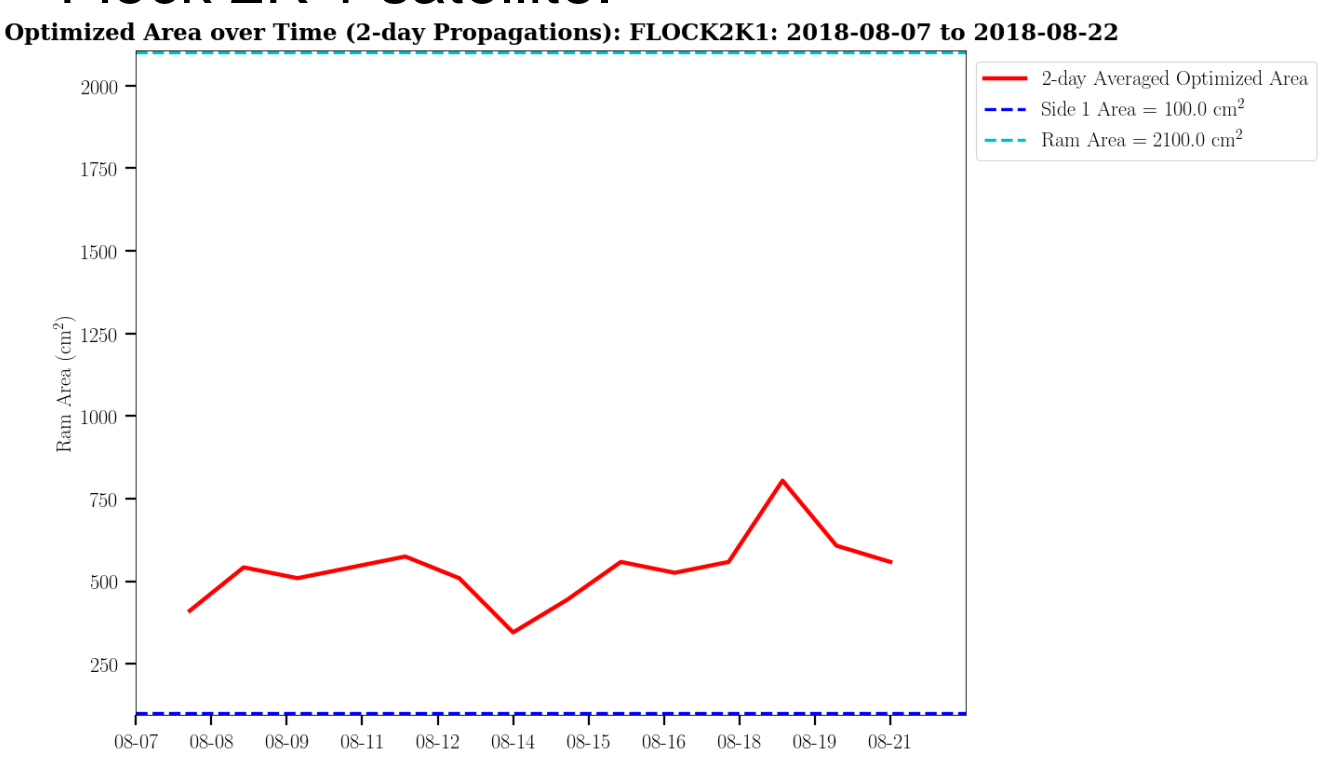


Figure 8: AROPT's resulting optimized areas for Flock 2K 1 during quiet time.

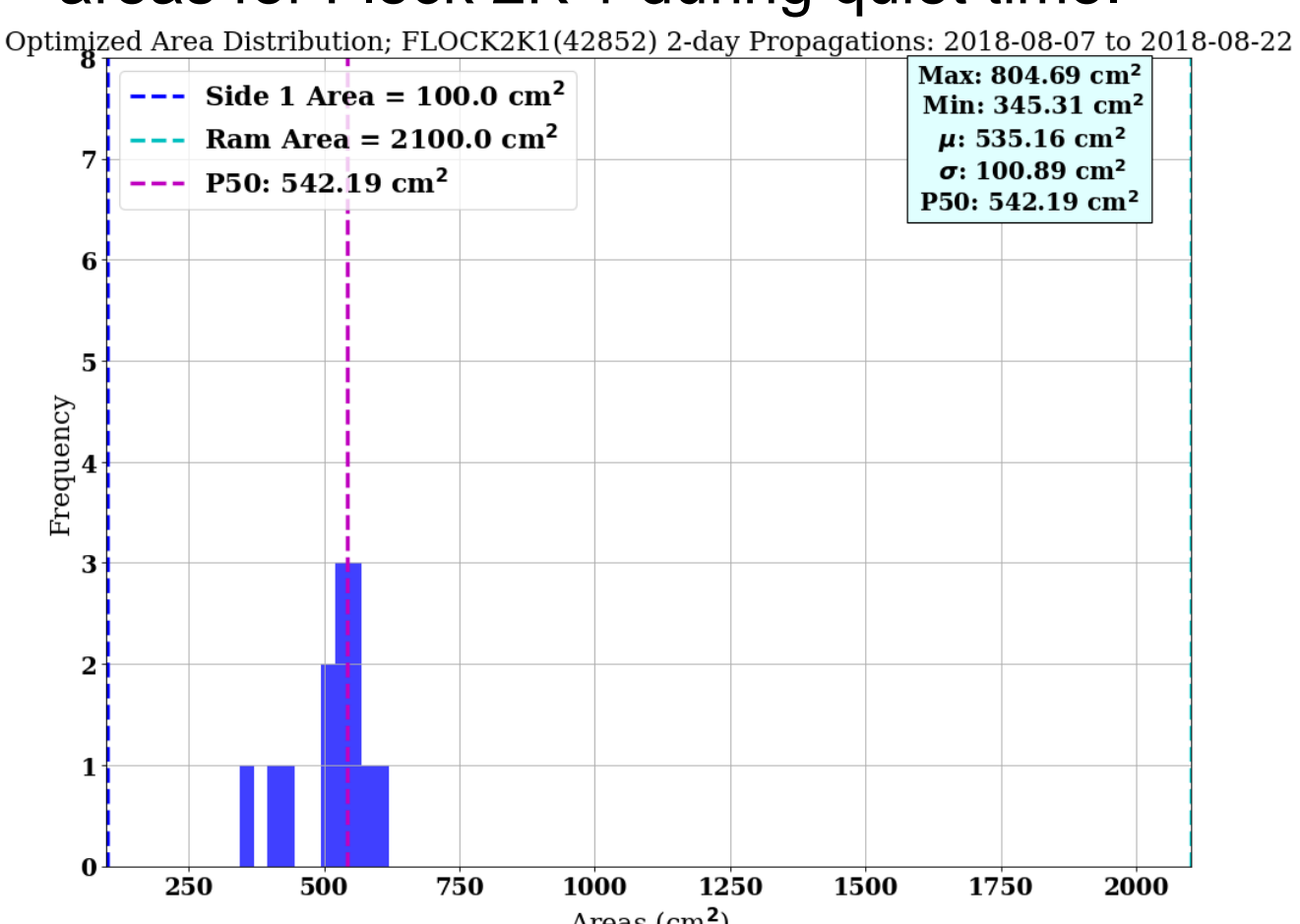


Figure 9: Flock 2K 1 optimized areas during quiet time assembled into a histogram.

## RESULTS

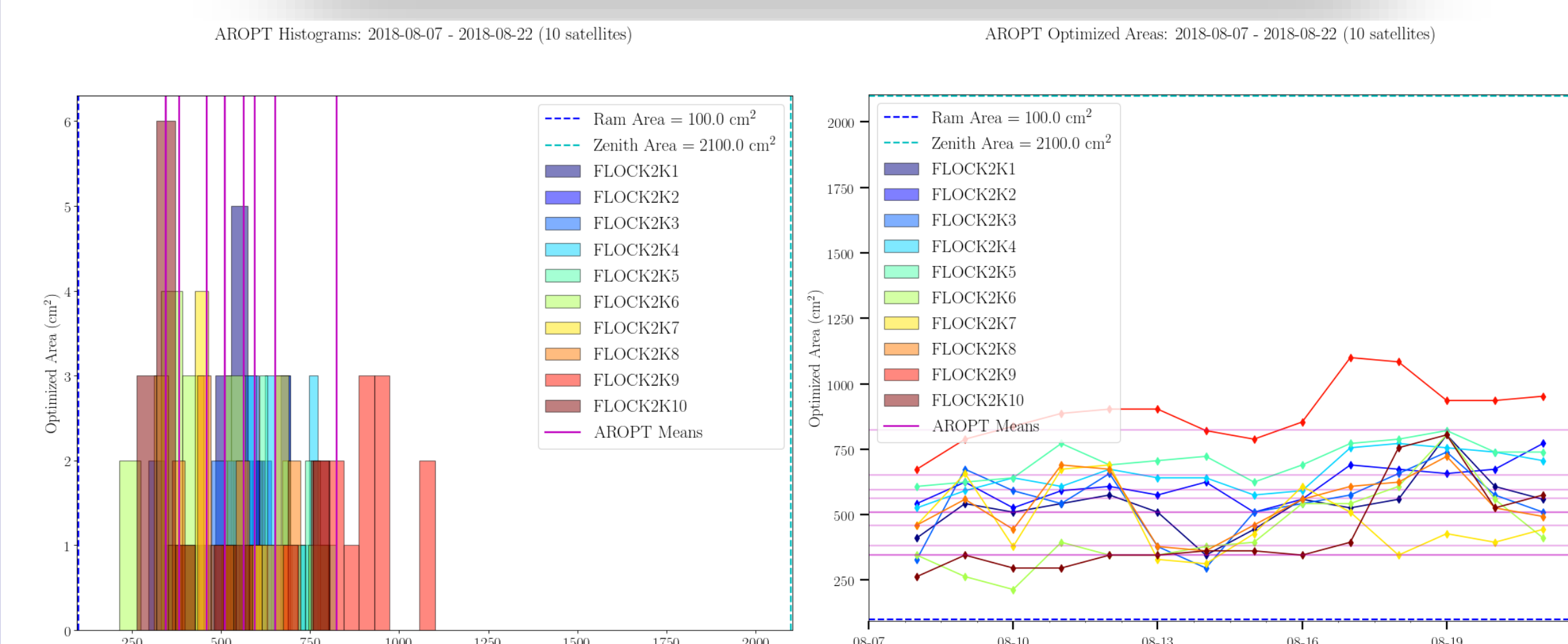


Figure 10: Overlapping histograms of optimized area distributions for all Flock 2K satellites (left) and their corresponding optimized areas over time (right).

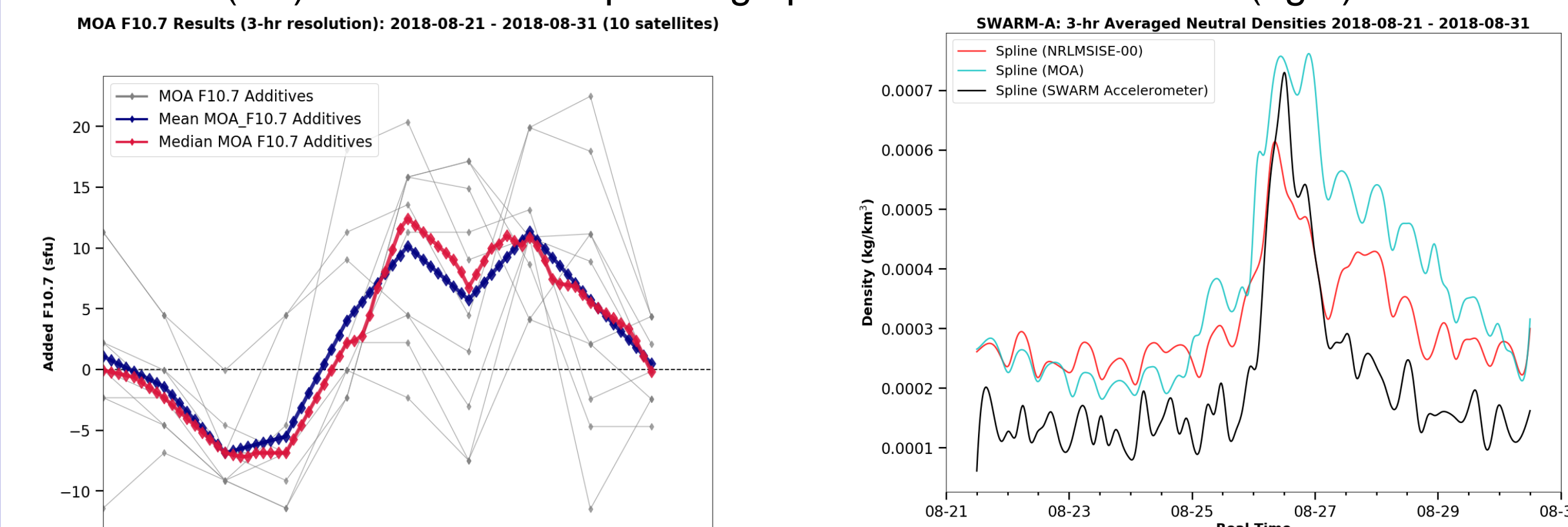


Figure 11: Linearly-interpolated F10.7 corrections across all 10 satellites.

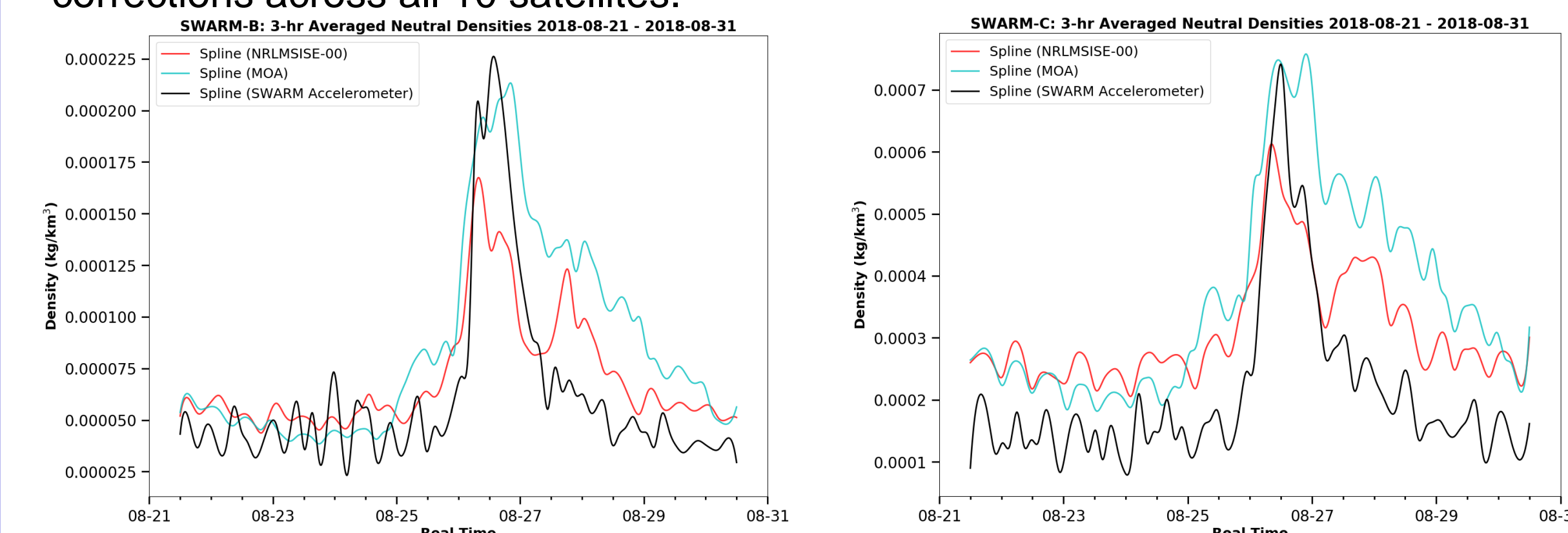


Figure 12: Corrected densities along-track SWARM-A.

Figure 13: Corrected densities along-track SWARM-B.

Figure 14: Corrected densities along-track SWARM-C.

## DISCUSSION AND FUTURE WORK

- Density correction allowing for reductions up to 10% error in magnitude, on par with results obtained by Doornbos et al. 2008.
- Post-main phase corrected densities are consistently too large.
- Densities more accurate than NRLMSISE-00 during initial onset.
- Raw corrected densities exhibit much greater amplitude variation compared to SWARM.
- Must validate the algorithm in other storms of varying intensities and using a variety of other calibration targets.
- Incorporate the third sub-process of correcting 3-hr  $a_p$  during storm main phase.

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## ACKNOWLEDGEMENTS

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