Generalized Time-Series Analysis for In-Situ Spacecraft Observations: Anomaly Detection and Data Prioritization using Principal Components Analysis and Unsupervised Clustering

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Abstract

In-situ spacecraft observations are critical to our study and understanding of the various phenomena that couple mass, momentum, and energy throughout near-Earth space and beyond. However, on-orbit telemetry constraints can severely limit the capability of spacecraft to transmit high-cadence data, and missions are often only able to telemeter a small percentage of their captured data at full rate. This presents a programmatic need to prioritize intervals with the highest probability of enabling the mission's science goals. Larger missions such as the Magnetospheric Multiscale mission (MMS) aim to solve this problem with a Scientist-In-The-Loop (SITL), where a domain expert flags intervals of time with potentially interesting data for high-cadence data downlink and subsequent study. Although suitable for some missions, the SITL solution is not always feasible, especially for low-cost missions such as CubeSats and NanoSats. This manuscript presents a generalizable method for the detection of anomalous data points in spacecraft observations, enabling rapid data prioritization without substantial computational overhead or the need for additional infrastructure on the ground. Specifically, Principal Components Analysis and One-Class Support Vector Machines are used to generate an alternative representation of the data and provide an indication, for each point, of the data's potential for scientific utility. The technique's performance and generalizability is demonstrated through application to intervals of observations, including magnetic field data and plasma moments, from the CASSIOPE e-POP/Swarm-Echo and MMS missions. Generalized Time-Series Analysis for In-Situ Spacecraft Observations:
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12 Key Points:

- Spacecraft often cannot transmit all measurements to Earth at full cadence due to
 telemetry bandwidth limitations.
- Many missions must implement complex data prioritization schemes to ensure only the
 highest-priority data is transmitted at high cadence.
- The proposed data prioritization technique is highly generic, compatible with inexpensive
 hardware, and suitable for low-cost missions.
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20 Abstract

In-situ spacecraft observations are critical to our study and understanding of the various 21 phenomena that couple mass, momentum, and energy throughout near-Earth space and beyond. 22 However, on-orbit telemetry constraints can severely limit the capability of spacecraft to transmit 23 high-cadence data, and missions are often only able to telemeter a small percentage of their 24 25 captured data at full rate. This presents a programmatic need to prioritize intervals with the highest probability of enabling the mission's science goals. Larger missions such as the Magnetospheric 26 Multiscale mission (MMS) aim to solve this problem with a Scientist-In-The-Loop (SITL), where 27 a domain expert flags intervals of time with potentially interesting data for high-cadence data 28 downlink and subsequent study. Although suitable for some missions, the SITL solution is not 29 always feasible, especially for low-cost missions such as CubeSats and NanoSats. This manuscript 30 presents a generalizable method for the detection of anomalous data points in spacecraft 31 observations, enabling rapid data prioritization without substantial computational overhead or the 32 need for additional infrastructure on the ground. Specifically, Principal Components Analysis and 33 One-Class Support Vector Machines are used to generate an alternative representation of the data 34 and provide an indication, for each point, of the data's potential for scientific utility. The 35 technique's performance and generalizability is demonstrated through application to intervals of 36 observations, including magnetic field data and plasma moments, from the CASSIOPE e-37 38 POP/Swarm-Echo and MMS missions.

39 Plain Language Summary

Measurements captured by spacecraft are necessary to our understanding the space environment 40 near Earth and throughout our solar system. However, spacecraft can often only transmit a small 41 portion of the data they capture back to Earth. This means that many spacecraft must prioritize 42 intervals of data that have the highest probability of helping to further our understanding of these 43 environments. Some missions utilize humans, on Earth, to help select these scientifically important 44 intervals. This solution, called the Scientist-In-The-Loop, can be too expensive or 45 programmatically complex for many small missions to implement. This manuscript presents a 46 technique for the detection of anomalous events in spaceflight measurements using statistical 47 analysis and machine learning. These detected anomalies can be used to prioritize data that has a 48 high probability of scientific relevance. Further, the proposed technique is highly generalizable 49 and computationally lightweight, making it suitable for a variety of missions. Several case studies 50 from multiple existing missions will be analyzed throughout this paper. 51

52 **1 Introduction**

Magnetic field sensors are one of the many science instruments that have been a 53 fundamental part of space exploration since its inception. Some of the first satellites, such as the 54 late-1950's Sputnik 3 and Explorer 6, carried fluxgate magnetometers to collect scientific data 55 (Gordon & Brown, 1972). Since then, the science of magnetometry and spaceflight has been 56 57 advanced such that we can make magnetic field measurements of far-flung bodies such as asteroids (Weiss et al., 2023), Mars (Connerney et al., 2015), Jupiter (Connerney et al., 2017), and the Sun 58 (Bale et al., 2016). The need to understand fundamental physical processes in space, such as 59 magnetic reconnection, has driven requirements for the telemetry of measurements at higher and 60 higher cadences (Phan et al., 2016). Additionally, our desire to enable comprehensive 61 understanding, nowcasting, and forecasting of Earth's near-space environment has led to the 62

development of the inexpensive CubeSat form factor and a greater number of multi-spacecraft missions. CubeSats allow cost effective proliferation of measurements that can vastly improve our models through data assimilation and machine learning techniques. The last several years have seen the emergence of large constellations, leading to a multitude of challenges, particularly in the handling and telemetry of the massive quantities of raw data available onboard each spacecraft (Liddle et al., 2020; Zhan et al., 2020).

Many missions, due to programmatic constraints on telemetry rates, are unfortunately 69 unable to downlink all of their captured data to Earth for analysis. Instead, mission operators and 70 science teams must make decisions about which intervals of time to transmit high-cadence data 71 (i.e., burst data). Some missions will transmit a lower-cadence data product (i.e., survey data) 72 during intervals deemed less important (Lepping et al., 1995), and some missions will simply not 73 telemeter these intervals (Yau & James, 2015). Intervals of burst data to be telemetered are 74 typically determined by estimated spacecraft position, by humans on the ground diligently 75 monitoring low-cadence data, or by carefully calibrated onboard algorithms which search the high-76 cadence data for mission-specific triggers. 77

The Magnetospheric Multiscale (MMS) mission (Burch et al., 2016) utilizes a particularly 78 thorough approach to the identification and prioritization of burst data to telemeter to the ground 79 (Baker et al., 2016). During spacecraft traversal through predetermined regions of interest, the 80 MMS instruments always capture data at their high cadence burst rates. This high-cadence data is 81 stored onboard while the lower-cadence survey data is telemetered to the ground and analyzed. If 82 83 the survey data shows potentially interesting phenomena, short intervals of burst data can be downlinked from the spacecraft. The MMS mission prioritizes burst data using two techniques: 84 85 the Scientist in the Loop (SITL) and the Automated Burst System (ABS).

The ABS, as its name indicates, runs automatically onboard each spacecraft and provides 86 a data ranking metric to be downlinked alongside the survey data (Baker et al., 2016). This system 87 uses data quality indicators calculated by each instrument to rank the available burst data in a 88 downlink prioritization queue. The last item in the queue will be the first to be overwritten should 89 a higher-ranking interval be identified. Although 34 data quality indicators are available for burst 90 91 triggering, the early mission used only large gradients in the Z_{GSM}-component of the measured magnetic field to prioritize burst intervals while the data quality indicators were characterized. 92 After two years of careful parameterization, ~6 data quality indicators are now used by the ABS 93 94 for prioritization of burst data containing mission-specific phenomena of interest (Argall et al., 2020). 95

The SITL is a manual option that can validate or override the selections made by the ABS. A domain expert – with access to MMS survey data, spacecraft-calculated trigger metadata, and data from other satellites or ground systems – determines the priority of data for downlink using specialized software, ensuring that data with high scientific significance is telemetered (Baker et al., 2016).

Both the ABS and SITL burst selection schemes for data prioritization require substantial scientific infrastructure and potentially costly overhead in their implementation. Extensive onboard triggering logic, years of parameter characterization, dedicated interval labeling time from experts during in-situ mission operations, and bespoke burst prioritization software are almost certainly infeasible for low-budget missions. One method has been recently proposed which aims to reduce the need for infrastructureintensive SITL activities on MMS using supervised machine learning (Argall et al., 2020). Although their technique shows excellent results and enables a great reduction in the reliance on manual labeling tasks, it requires a large set of expert-labeled data during the training of their segmentation network and therefore does not eliminate the need for the SITL infrastructure entirely.

This manuscript proposes the use of a common dimensionality reduction technique, 112 coupled with unsupervised clustering, to provide a robust and generalizable method for detecting 113 anomalous intervals of time series spacecraft observations. This method is intended for use as a 114 component of a drop-in burst data prioritization system for missions where the infrastructure and 115 cost associated with more sophisticated and mission-specific solutions are not feasible. Although 116 the SpaceX Starlink constellation currently dominates the Low Earth Orbit environment with over 117 2,000 satellites currently in orbit and approval granted for 12,000 total satellites, it is unlikely to 118 remain the only major constellation in orbit (Ma et al., 2023; McDowell, 2020). These mega-119 constellations are a heavy burden on ground systems, requiring complex protocols for dealing with 120 telemetry, command, and tracking (Berner, 2019). The proposed tool would be an invaluable asset 121 for such constellation missions, enabling a higher degree of distributed autonomy in their space 122 operations. 123

The following sections of this manuscript describe the proposed technique and demonstrate its performance with several case studies on observational measurements obtained from the CASSIOPE e-POP/Swarm-Echo spacecraft and one of the Magnetospheric Multiscale mission satellites.

128 2 Methodology

129 2.1 Dimensionality Reduction via Principal Components Analysis

130 Principal Components Analysis (PCA) is one of the oldest and most popular multivariate

statistical analysis techniques used to reduce the dimensionality of large datasets (Jolliffe &

132 Cadima, 2016). Mathematically, PCA is performed by identifying the eigenvectors of the

133 covariance matrix associated with the data matrix under observation (X) via

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$\boldsymbol{X}^{T}\boldsymbol{X} = \boldsymbol{V}\boldsymbol{\Lambda}\boldsymbol{V}^{T} \tag{1}$

where the columns of *V* correspond to the eigenvectors and the diagonal elements of Λ are the associated eigenvalues. For convenience, let each eigenvector V_i be ordered by the magnitude of its associated eigenvalue.

138 The projection and subsequent dimensionality reduction can be realized through

$$\boldsymbol{P} = \boldsymbol{X} \boldsymbol{V}_{\boldsymbol{q}} \tag{2}$$

where V_q is a matrix whose columns are only the first q eigenvectors from V. Throughout this manuscript, the dimensionality of the output projection is fixed to two (i.e., q = 2) in order to reduce the computational complexity associated with the analysis of the projected data.

143 The specific data matrix being analyzed in this manuscript is generated by concatenating 144 *R* consecutive time intervals, of length *L*, from the original time series (x, with length *N*) into a 145 single matrix via 146

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$$\mathbf{X} = \begin{bmatrix} x(1) & x(L+1) & \dots & x(RL-L+1) \\ \dots & \dots & \dots & \dots \\ x(L) & x(2L) & \dots & x(RL) \end{bmatrix}.$$
 (3)

147 It should be noted that X is a reduced version of the trajectory matrix passed into PCA by the 148 Singular Spectrum Analysis (SSA) technique (Finley et al., 2023). Both SSA's trajectory matrix 149 and the data matrix used throughout this paper are constructed to enable information about the 150 temporal variation of a signal to be utilized. The reduced trajectory matrix (i.e., no overlapping 151 samples in each consecutive column) was used here to reduce the computational complexity 152 associated with the overall algorithm.

An example of this variation on PCA, applied to one axis of the vector magnetometer data captured by the CASSIOPE e-POP/Swarm-Echo magnetic field instrument (Wallis et al., 2015; Yau & James, 2015), is shown in Fig. 1. Figure 1(a) illustrates the 2.5-minute interval of data, at a sampling rate of 160 Hz, to be processed. Figure 2(b) shows the two-dimensional representation, given by P, of the data following the application of PCA (with q = 2) on a data matrix constructed from 0.5-second consecutive intervals taken from the signal in Fig. 1(a).



Figure 1: Dimensionality reduction via application of PCA on concatenated 0.5-second intervals of magnetic field data. (a)
 Inboard magnetometer data from CASSIOPE e-POP/Swarm-Echo MGF; (b) Two-dimensional representation of the 0.5-second
 intervals of (a) following PCA while retaining only two principal components.

Each point in Fig. 1(b) is a projection of one of the 0.5-second intervals of Fig. 1(a). It can 163 be seen that many of these points are clustered tightly near the origin, whereas some of the points 164 are outlying near the periphery. This implies that the majority of the time intervals exhibit similar 165 behavior when represented using only the first two principal components (i.e., those that describe 166 167 the largest variance in the original data matrix). However, some intervals show very different behavior in terms of these maximum-variance components. Automatic clustering of this two-168 dimensional representation should reveal anomalous time intervals in the original signal and is 169 discussed in detail in Sec. 2.2. 170

171 2.2 Clustering with One-Class Support Vector Machine

Machine learning techniques have become increasingly popular in the various space 172 physics research domains. Successful application of these techniques has been seen in methods for 173 auroral image classification (Clausen & Nickisch, 2018), recreating magnetohydrodynamic 174 environments from sparse sample spaces (Bard & Dorelli, 2021), space weather forecasting 175 (Camporeale, 2019), and many others. However, machine learning models can often be complex, 176 requiring large quantities of training data and computational resources. Once trained, these large 177 and complex models are often treated as 'black boxes,' and can lack interpretability (Angelov et 178 al., 2021). To increase potential applicability to low-cost and in-situ spaceflight hardware, a 179

machine learning-based clustering solution that is computationally efficient and easily understoodmust be utilized instead of a more complex model.

Support Vector Machines (SVMs) are a popular means of performing classification tasks 182 throughout a variety of fields including the biomedical sciences (Zhou et al., 2005) and industrial 183 engineering (Shin et al., 2005). This data labeling technique has seen widespread adoption due to 184 its high degree of robustness and interpretability (Hearst et al., 1998). Traditional SVMs are trained 185 by first projecting the labeled training data to a higher dimension feature space using a user-186 selected *kernel*. Next, a hyperplane that best separates the classes is calculated, although a slack 187 parameter is considered in this optimization. This slack parameter enables the trained SVM to 188 handle a small number of data points that cannot be separated using a hyperplane in the higher-189 190 dimension feature space, which is a common situation in real-world datasets (Noble, 2006). This trained SVM can now be used to classify new data not seen during the training process. 191

Slight modification of the traditional SVM framework leads to a technique known as the 192 One-Class Support Vector Machine (OC-SVM), a common unsupervised approach to data 193 classification and anomaly detection (Yin et al., 2014). These OC-SVMs operate in a similar 194 fashion to the traditional SVM but calculate a hyperplane that optimally separates the data from 195 the origin, not by separating pre-labeled classes (Amer et al., 2013). Here, the primary user-defined 196 control parameter is v, which lies in the range (0,1] and determines the upper bound on the number 197 of allowed errors and a lower bound on the number of data points used when calculating the 198 199 separating hyperplane (Chang & Lin, 2001).



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201 Figure 2: One-Class Support Vector Machine clustering applied to the reduced-dimension data illustrated in Fig. 1.

Figure 2 illustrates the result of passing the two-dimensional output of PCA, shown in Fig. 1(b), through an OC-SVM. This OC-SVM was trained with a Gaussian kernel and a v-value of 0.3. Points shown in red were those that were considered anomalous, whereas the points shown in blue were considered nominal. Since each point in Fig. 2 represents a 0.5-second interval of the original input shown in Fig. 1(a), the associated labels can be directly applied to each interval in the original input time series. The result of this inversion procedure, and additional examples, will be discussed in detail in Sec. 4.

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210 3 Data and Preprocessing

211 3.1 CASSIOPE e-POP/Swarm-Echo MGF

One primary source of the data analyzed in previous sections and throughout the remainder 212 of this manuscript is the CASSIOPE/Swarm-Echo Magnetic Field instrument (Wallis et al., 2015; 213 Yau & James, 2015). The Magnetic Field instrument (MGF) consists of a pair of identical fluxgate 214 magnetometers mounted inline on a single boom at approximately 0.6 m and 0.9 m from the body 215 216 of the spacecraft. Both magnetometers capture the local magnetic field at a rate of 160 Hz and downlink the data when telemetry constraints allow. Although the magnetometer pair (i.e., 217 gradiometer) could be used to mitigate local interference from the host spacecraft and improve 218 data fidelity (Finley et al., 2023), the measurements used in this manuscript were taken from only 219 the inboard magnetometer mounted closer to the spacecraft. This provides some insight into how 220 the proposed anomaly detection technique handles data contaminated by local interference, which 221 222 is almost constantly observed at the magnetometers as high-frequency oscillations caused primarily by the spacecraft's attitude control systems (i.e., reaction wheels). The magnetometer 223 measurements used in this manuscript have had a near-DC baseline removed using a 20-s moving 224 average prior to analysis and visualization. 225

226 3.2 Magnetospheric Multiscale Mission FGM

Another source of magnetic field data used in the remaining sections of this manuscript is 227 the Magnetospheric Multiscale mission (MMS) Magnetometers (Burch et al., 2016; Russell et al., 228 2016). Although data is available from all four MMS satellites, only data from MMS1 was utilized 229 throughout this manuscript. The MMS Magnetometers consist of a near-identical pair of fluxgate 230 sensors, with each sensor mounted at the end of two separate 5-meter booms. The measured data 231 is reported as a high-fidelity triaxial vector data product (called FGM) with three possible sampling 232 rates: slow survey at 8 Hz; fast survey at 16 Hz; and, burst data at 128 Hz. In this manuscript, only 233 fast survey and slow survey data were utilized. If both fast and slow survey data were present in 234 the time interval to be analyzed, the data was resampled to match the slow survey data rate for 235 consistency of analysis. As with the CASSIOPE data discussed in Sec. 3.1., a 20-s moving average 236 has been removed from the MMS data prior to analysis and visualization. 237

238 3.3 Magnetospheric Multiscale Mission FPI

An additional set of data used in the remaining sections of this manuscript is the MMS Fast 239 240 Plasma Investigation (Burch et al., 2016; Pollock et al., 2016). As with the magnetic field data, only data from the MMS1 satellite was used in this manuscript. The Fast Plasma Investigation 241 (FPI) for MMS comprises multiple top-hat electrostatic analyzers (Carlson et al., 1982) to 242 determine in situ the fluxes of electrons and ions as functions of energy and direction. The FPI, its 243 measurements, and methods of computation are described in detail in (Pollock et al., 2016). FPI 244 acquires a full 3D set of electron samples (32 energies \times 32 azimuths \times 16 polar sections) every 245 30 ms, and an equivalent set of ion samples every 150 ms. When telemetered to the ground, these 246 samples are employed to compute electron and ion fluid parameters such as number density, bulk-247 flow velocity, pressure, and others. Each parameter is computed as a summation of fluxes weighted 248 249 by appropriate physical factors. These parameters are essential quantities required for many scientific analyses of space plasmas, for example in comparisons of observations with the output 250 of magnetohydrodynamic (MHD) simulations that predict the overall behavior of plasma as a fluid. 251

The ion parameters (i.e., number density and velocity) shown in this manuscript are reported at the Fast Survey rate of ~0.22 Hz (i.e., captured at full cadence and averaged once every 4.5 s).

For the FPI, approximate fluid-like parameters are also computed in a simplified fashion by the instrument processor onboard the spacecraft. The computation is approximate because the summations are not weighted by the proper factors necessary to obtain true physical quantities. However, it has been shown that these onboard quantities, known as pseudo-moments, can be rescaled to serve as proxies for the true physical parameters (Barrie et al., 2018). As a result, analysis of the derived plasma moments using the proposed anomaly detection technique is indicative of the technique's performance when applied to the onboard pseudo-moments.

261 3.3 Magnetospheric Multiscale Mission SITL

To verify the capability of the proposed technique to identify intervals of time relevant to 262 the MMS mission's science objectives, this manuscript also includes data from the MMS Scientist-263 in-the-Loop (SITL) report. MMS is the first mission with both the spatial and temporal resolution 264 to resolve electron-scale dynamics. This requires MMS to capture much more data than it can 265 telemeter to the ground. As discussed in Sec. 1, this has led to the development of MMS' burst 266 management system, consisting of the Automated Burst System (ABS), Scientist-in-the-Loop 267 (SITL), and the Ground Loop System (GLS). The SITL, specifically, is a role passed among MMS 268 team member volunteers who search through the survey data to identify and select time intervals 269 that may contain relevant events for burst-mode downlink. Survey data has insufficient resolution 270 to capture electron-scale dynamics, so the SITL must over-select events to ensure that mission-271 272 critical science data is captured. The SITL is guided by mission-level science objectives that enable assignment of a figure of merit (FOM) value to each interval examined (Argall et al., 2020; 273 Hasegawa et al., 2023; Phan et al., 2016). These FOMs subsequently inform the mission of the 274 highest-priority data to be selected and downlinked at burst rate. Each SITL selection is 275 accompanied by a short description that is searchable and parsable. These descriptions have 276 previously been used to train a supervised machine learning model that uses the SITL report to 277 make future predictions about which data should be selected for downlink (Argall et al., 2020). 278 This model is installed in the near real-time data processing GLS so that predictions can be made 279 280 as soon as the preliminary low-cadence data is downlinked in order to guide the SITL (along with the ABS selections). The SITL data itself is not publicly available, as it is not considered science-281 quality, but the reports describing the selected events and their time range can be searched through 282 the MMS Mission Events webpage (https://lasp.colorado.edu/mms/sdc/public/about/events/#/). 283 Additional tools, such as PyMMS, have been developed that enable rapid searching of these reports 284 (Argall et al., 2022). 285

286 **4 Results**

287 4.1 Observations of Current Sheets with Embedded Alfvén Waves (CASSIOPE)

288 Section 2 illustrated (through Fig. 1 and Fig. 2, respectively) the dimensionality reduction 289 and unsupervised clustering techniques utilized in the proposed method of automated anomaly 290 detection by applying them to a short interval of magnetic field data captured by the CASSIOPE 291 e-POP/Swarm-Echo inboard magnetometer. Figure 3(a) shows the proposed method's output after 292 inverting the labels given to each point shown in Fig. 2 back onto the original magnetometer data 293 shown in Fig. 1(a), as described in Sec. 2.2. Points in blue are those considered nominal by the 294 technique, whereas points labeled red have been flagged as anomalous or potentially scientifically

relevant. Figure 3(b) provides the spectral content for the interval shown in Fig. 3(a). The spectral 295 content and time series for this 2.5-minute interval shows previously identified Alfvénic activity 296 embedded in a large current sheet at roughly 6:48:55 UTC (Miles et al., 2018), with a smaller 297 current sheet observed near 6:48:10 UTC. The current sheet containing embedded Alfvénic 298 activity is highlighted in gray in Fig. 3(a). It can be seen that the proposed method of anomaly 299 detection was able to accurately identify these interesting intervals containing broadband magnetic 300 activity, even in the presence of the potentially obfuscating reaction wheel interference observed 301 at ~15 Hz. 302





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308 4.2 Observations of Spacecraft Maneuvers (CASSIOPE)

The next interval of data analyzed in this section is a 20-minute interval of CASSIOPE e-309 POP/Swarm-Echo magnetometer data captured when the spacecraft was performing maneuvers. 310 As a result, the spacecraft's four reaction wheels were changing their spin frequency rapidly during 311 this period. Figure 4(a) shows the labeled time series output by the method with a window length 312 of five seconds. Anomalous intervals are shown in red, normal points are shown in blue. Figure 313 314 4(b) shows the spectral content of the interval shown in Fig. 4(a). Highly dynamic high-frequency activity resulting from the reaction wheels' diverging operational rates can be seen for a five-315 minute interval starting at approximately 4:55:00 UTC, with other low-frequency perturbations 316 occurring throughout the total 20-minute interval. It can be seen that the detected anomalies 317 directly correspond to changes in behavior in the measured magnetic field due to the spacecraft 318 maneuvers. This example helps to both illustrate the ability of the proposed technique to 319 simultaneously identify dynamic behavior in both high- and low-frequency bands, as well as 320 identify intervals critical to spacecraft operations (i.e., spacecraft maneuvers). 321



Figure 4: Proposed method of anomaly detection applied to interval of CASSIOPE e-POP/Swarm Echo MGF data containing
 highly dynamic high-frequency signatures caused by spacecraft reaction wheels. (a) Time series with anomalous intervals plotted
 in red, nominal in blue. (b) Spectrogram associated with the data in (a).

327 4.3 Observations of Interplanetary Shocks and EMIC Waves (MMS)

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The next period of data analyzed in this manuscript is a full 24 hours of magnetic field data 328 captured by the MMS magnetometer suite in December 2015. Figure 5(a) shows the result of the 329 proposed method of anomaly detection applied to the full day of data with a window length (L) of 330 5 minutes. As in the previous examples, points shown in red correspond to detected anomalies and 331 332 points shown in blue have been identified as nominal. Figure 5(b) shows the spectrogram associated with the magnetic field measurements in Fig. 5(a). Note that this data has been 333 resampled to match the slowest sampling rate present in the interval (i.e., 8 Hz), as described in 334 Sec. 3.2. The frequent broadband magnetic phenomena seen throughout the first ~40% of this 24-335 hour interval is attributed, by the MMS SITL, to observations of the bow shock and magnetopause. 336 The broadband magnetic activity seen in the last ~15% of the day was similarly reported by SITL 337 to correspond to observations of the magnetopause. The large-amplitude (i.e., > 200 nT) 338 phenomena at approximately 17:00 UTC corresponds to the perigee of the MMS1 spacecraft. This 339 day of data, which was analyzed extensively by Engebretson et al. (2018), also contains MMS' 340 observations prior to, during, and after an interplanetary shock. The resultant compression of the 341 magnetosheath was observed by MMS at approximately 13:24 UTC, with structured EMIC wave 342 activity occurring before and after. The gray highlighted region in Fig. 5(a) corresponds to the 343 region of interest (from 13:00-14:00 UTC) containing the majority of this activity (Engebretson et 344 al., 2018). It can be seen that the proposed method of anomaly detection is able to accurately label 345 a significant portion of the interesting magnetic field data occurring at bow shock and 346 magnetopause crossings before noon and near midnight. In addition, the technique also labels 347 several of the large-amplitude perturbations in the magnetic field measurement near and during 348 the 13:00-14:00 UTC region of interest. Figures 6(a) and 6(b) provide a zoomed view of this 349

region's time series and spectral content, respectively. No change has been made to the anomaly 350 labeling, which still corresponds to the result when the proposed method is applied to the full 24-351 hour period. It is important to note that although the proposed technique has successfully identified 352 the broadband signal corresponding to the magnetosheath compression, much of the structured 353 EMIC wave activity occurring during this region of interest has not been labeled as anomalous. 354 This result is obtained due to the input parameters selected for this example. The next example, 355 shown in Fig. 7, will illustrate the effect of varying the input parameters when the anomaly 356 detection technique is applied. 357



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Figure 5: Anomaly detection technique applied to a 24-hour interval of MMS FGM data containing a variety of scientifically
 interesting phenomena including observations of the bow shock, magnetopause, and compressions of the magnetopause due to an
 interplanetary shock. (a) Time series FGM measurements with anomalous intervals identified by the proposed technique plotted
 in red, nominal intervals plotted in blue. (b) Spectrogram associated with the data shown in (a).



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Figure 7 illustrates the impact of changing the signal length (N) and window length (L) on 368 the performance of this technique. Specifically, Figure 7(a) shows the labeling output by the 369 proposed method of anomaly detection when applied to only the 13:00-14:00 UTC region of 370 interest highlighted in Fig. 5(a) and shown in Fig. 6(a). Note that this impacts the signal length, N, 371 without changing the window length, L. Although more of the structured wave activity in this 372 region is correctly labeled as anomalous, this result may still not be sufficient for some 373 applications. Reducing the window length (L) from five minutes to one second, as seen in Fig. 374 7(b), provides a more detailed labeling of the anomalous samples within the total interval. As a 375 result, much more of the structured EMIC wave activity occurring after the magnetosheath 376 incursion is identified as anomalous. Although the parameter space inherent to the proposed 377 method of anomaly detection is small, these results clearly show the relevance of the input 378 parameters to the method's result. Selection of these parameters, and their effect on the method's 379 computational complexity, will be discussed in greater detail in Sec. 5.1. 380



Figure 7: Illustration of the impact of the user-defined input parameters on the output of the proposed anomaly detection, shown for the same data seen in Fig. 6. (a) Time series output by the anomaly detection technique for a signal length of one hour and a window length of five minutes. (b) Time series output by the anomaly detection technique for a signal length of one hour and a window length of one second.

386 4.4 Observations of Magnetopause Crossings and the Bow Shock (MMS)

The next experiment in this manuscript demonstrates the applicability of the proposed 387 anomaly detection technique to spacecraft observations other than magnetic field measurements. 388 Specifically, the technique is applied to a large interval of magnetic field data, as well as ion 389 number density and ion velocity, as measured by the MMS spacecraft on 15 May 2023. Figure 390 8(a) shows a 24-hour period of magnetic field measurements, Fig. 8(b) and Fig. 8(c) show the ion 391 number density and ion velocity corresponding to the same period. The points illustrated in red are 392 those considered anomalous by the proposed technique when it was applied to the total interval, 393 whereas points shown in blue are considered nominal. It should be noted that the last 3.75 hours 394 of ion measurements were not available because the spacecraft transitioned from Fast Survey to 395 Slow Survey rate, and the FPI does not operate during Slow Survey mode. This period of missing 396 data has been padded with nominal-labeled zeroes, after the anomaly detection algorithm was 397 applied, for consistency with the magnetic field data. The gray region highlighted in Fig. 8 398 corresponds to a region of data containing various physical phenomena identified by the MMS 399 SITL, such as observations of the bow shock, magnetopause, and boundary layer crossings. These 400 SITL identifications, including the time range and the associated Figure of Merit (FOM) are 401 detailed in Tab. 1. A higher FOM is assigned to a selection that should be downlinked at higher 402 403 priority, with a maximum value of 199 being assigned to typical events, while higher FOMs are reserved for calibration intervals or other special events (Argall et al., 2020). 404

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408 Table 1: MMS SITL report for 15 May 2023. Columns 1-2 provide the start and stop time for each interval of interest. Column 3
 409 shows the Figure of Merit (FOM) associated with each interval. Column 4 shows SITL remarks for each interval.

Start Time (UTC)	Stop Time (UTC)	Figure of Merit (FOM)	Discussion
12:49:43	13:17:33	100	Partial Bow Shock
13:25:33	13:32:43	75	Partial Bow Shock
13:37:33	13:46:43	75	String of Partial Bow Shock Crossings
15:34:03	15:58:13	90	Partial Magnetopause
16:24:53	16:49:43	150	Full Medium Shear Magnetopause
17:51:43	17:56:13	55	Boundary Layer Traversal

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In addition to these SITL-identified phenomena, the magnetic field data shows frequent 411 broadband perturbations throughout the first half of the day, as well as a large data spike occurring 412 near midnight that corresponds to a perigee pass of the MMS satellite. It can be seen that, for all 413 three sets of input data, the anomaly detection technique successfully identifies much of the gray-414 highlighted region selected by the SITL as anomalous, but with a significant number of false 415 positives shown for the magnetic field data when compared to only the SITL selections. These 416 false positives in relationship to the SITL selection can be attributed to successful identification of 417 the broadband perturbations of the magnetic field during the first half of the day. This number of 418 false positives is greatly reduced in the labels associated with both the ion density and velocity, 419 implying that a combination of parameters classified by the proposed technique may be used in 420 the identification of only phenomena similar to those prioritized by the MMS SITL. 421



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Figure 9 provides a zoomed view of only the gray-highlighted region shown in Fig. 8. As 427 before, Fig. 9(a) - Fig. 9(c) show the magnetic field, the ion number density, and the ion velocity 428 429 measured by MMS. The nominal and anomalous labels assigned by the proposed anomaly detection technique correspond to the blue and red points, respectively. These labels are identical 430 to the labels seen in Fig. 8; only the time scale of the plot has been changed to visualize the 431 technique's output more clearly during the intervals selected by SITL. The regions highlighted in 432 gray in Fig. 9 correspond to the specific time periods identified by SITL (described in Tab. 1) as 433 containing observations of the bow shock, magnetopause, and boundary layer crossings. In a 434 similar trend to the larger period shown in Fig. 8, the magnetic field data contains a greater number 435 of false positive identifications of anomalous events when compared to only the SITL selections, 436 and the ion moments, or combinations of the ion moments and magnetic field, may provide a 437 greater degree of accuracy for this case study. 438

The example shown in Fig. 8 and Fig. 9 clearly demonstrates the generic applicability of 439 the anomaly detection algorithm to a variety of spaceflight data products. Additionally, it shows 440 that greater utility may be leveraged from the proposed method by analyzing its output when 441 442 applied to several data products. For example, a weighting scheme generated by some logical operation of the binary labels (nominal or anomalous) associated with the method's output, when 443 applied to multiple data products, may provide a reduced number of potential false positives 444 depending on the target application. This is left as a potential topic for future study, and Section 5 445 will discuss several other avenues for future work utilizing this technique. 446 447



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 449</sup> Figure 9: Zoomed view of the region of interest shown in Fig. 8., with SITL-identified observations of the bow shock,
 450 magnetopause, and boundary layer crossings. (a) FGM magnetic field data. (b) FPI ion number density. (c) FPI ion velocity.

451 **5 Discussion & Future Work**

452 5.1 Parameter Selection & Computational Complexity

This manuscript has demonstrated the capabilities of the proposed method of anomaly detection to analyze data from a variety of instruments and successfully identify various interesting phenomena; however, the example shown in Fig. 5 through Fig. 7 has illustrated the need for appropriate parameter selection to enable the highest possible scientific return. The proposed technique has a small parameter space, with only three variables impacting the output result. Thus, each of these parameters plays a critical role in the outcome of the analysis, as well is its computational complexity.

The input signal length (N) determines the total length of data to be analyzed when 460 determining anomalous activity. The window length, L, determines the length of the consecutive 461 observations used as features when performing PCA. Finally, the tuning parameter v used in the 462 OC-SVM is, practically, the fraction of the consecutive observations that can be considered 463 anomalous when the data is clustered. In practice, this means that the input signal length (N)464 must be large enough that the majority of the data (i.e., at least the fraction given by $1 - \nu$) 465 should be considered nominal, based on the target application. The window length must also be 466 carefully considered when applying the technique as it corresponds to the scale on which 467 anomalies are detected. If large, slowly varying changes are to be identified, a longer window 468 length might be utilized; if small, rapid changes must be flagged, a shorter window length may 469

470 be more suitable.

Changes in these parameters also have a significant impact on the computational 471 complexity of the previously described technique. PCA, which is one of the fundamental 472 building blocks for this algorithm, has two basic steps: computation of the covariance matrix, 473 which has a complexity of $O(L^2R)$ for a given window length of L and a given number of total 474 consecutive observations, R; and, eigendecomposition of the covariance matrix, which has a 475 complexity of up to $O(R^3)$ in the worst-case scenario (Zhang et al., 2015). For the OC-SVM, the 476 most computationally expensive component is the model training, with a complexity of up to 477 $O(R^3)$. However, this complexity has been proven to be reduced by approximately one order of 478 magnitude using more complex techniques such as sequential minimal optimization (Kang et al., 479 2019). 480

It can be seen that the window length, the number of consecutive observation intervals, and the OC-SVM tuning parameter have a direct impact on the computational complexity, or the detected anomalies associated with the proposed algorithm. As a result, users must tune these parameters as appropriate to suite the capabilities of their hardware and the data being processed.

485 5.2 Potential for Implementation on Low-Cost Hardware

Although PCA has historically been one of the most popular statistical analysis techniques 486 used to reduce the dimensionality of large datasets (Jolliffe & Cadima, 2016), until recently 487 relatively few studies have evaluated the potential for PCA's implementation on embedded 488 hardware (Korat & Alimohammad, 2019). Early efforts to perform PCA on embedded hardware 489 relied on bespoke Very Large Scale Integration (VLSI) integrated circuits with complex 490 architecture (Tung-Chien Chen et al., 2008), but more recent works have leveraged modern and 491 relatively inexpensive FPGA technologies to perform either portions of the PCA computations 492 (Ali et al., 2013) or complete implementations of the PCA algorithm (Bravo et al., 2010; Korat & 493 Alimohammad, 2019). Additional research has proven that embedded implementation of PCA is 494 generic and highly scalable, enabling substantial improvements in the computational speed of the 495 technique across a range of applications (Shahrouzi & Perera, 2019). 496

Support Vector Machines (SVMs), which fall into the broad category of machine learning algorithms, have historically provided excellent performance when classifying complex and continuous features (Saidi et al., 2021). Generally, machine learning techniques are considered computationally expensive and challenging to implement on embedded hardware (Sze et al., 2017). However, several recent studies have shown the potential for the SVM algorithm to be implemented on a variety of embedded devices including VLSI integrated circuits and FPGAs (Amezzane et al., 2020; Loukrakpam & Choudhury, 2020).

In addition to being suitable for implementation on embedded hardware, both Principal Components Analysis and/or Support Vector Machines have been previously utilized for spacecraft fault detection and diagnosis (Yu Gao et al., 2012), the onboard detection of anomalous behavior in CubeSat solar panels (Cespedes et al., 2022), and other intelligent decision making applications onboard spacecraft (Jallad & Mohammed, 2014). This illustrates that not only can PCA and SVM be implemented in hardware, but that they have history in successful implementation for spaceflight applications.

511 5.3 Semantic Labeling of Prioritized Data

512 This manuscript has proposed a technique for the automated binary classification of time-513 series data as either anomalous (i.e., potentially of high scientific importance) or nominal, one 514 useful avenue of future work would be to provide additional semantic labels for the high-priority

data. These labels could indicate whether an identified event falls into a particular class of 515 geophysical event, such as shocks, magnetopause crossings, or whistler-mode waves. Several 516 machine learning techniques have been previously developed for the identification of events in 517 spaceflight data archives (Fordin et al., 2023; Vech & Malaspina, 2021), although the 518 computational intensity or large dimensionality of some of these algorithms make them potentially 519 unsuitable for deployment on spaceflight hardware. The proposed anomaly detection technique 520 provides utility to the pursuit of semantic labeling in two ways: firstly, as a low-cost data reduction 521 tool it can reduce the number of samples that must be processed by more complex algorithms, 522 decreasing the overall time complexity of the problem; secondly, the binary-labeled data can serve 523 as a powerful input feature vector if data reduction is not desired or required, potentially increasing 524 525 model performance.

526 5.4 Generalizability

527 This manuscript has illustrated the applicability of the proposed method of anomaly 528 detection to magnetic field data from MMS and CASSIOPE/Swarm-Echo, as well as the ion 529 density and velocity moments from MMS. Although this shows promise for the generalizability 530 of the technique to other platforms and other physical observations, thorough exploration of this 531 generalizability to other spacecraft observations (such as ion and electron pressures and 532 temperatures, as well as the housekeeping measurements critical to spacecraft operations) remains 533 an avenue for future work.

Testing the generalizability of the clustering model trained on a specific interval of time against different intervals is also an interesting future project. If the clustering learned by an OC-SVM trained on one interval of time (e.g., one of the 24-hour periods of MMS data) is applied to subsequent intervals (e.g., the next several days of MMS data) with meaningful results, the computational complexity of the proposed technique would decrease by eliminating the need to retrain a model for every interval under observation, enabling more rapid in-situ application.

540 6 Conclusions

The scientific measurements captured by in-situ spacecraft are critical to our study of the 541 physical phenomena that control the flow of mass, momentum, and energy throughout our solar 542 system. However, due to burdens on ground systems as a result of a rapid increase in the number 543 of active spaceflight missions and the ever-growing need for higher-cadence data, spacecraft are 544 often unable to transmit all of their data to Earth at full rate. As a result, missions must develop 545 techniques that enable prioritization of specific intervals with a high probability of importance to 546 their science goals. The techniques that have historically been used by large missions have 547 provided excellent results but come with high design and implementation costs, leaving them 548 potentially unsuitable for application on low-cost missions. This manuscript has proposed a 549 generic technique for the prioritization of data through the detection of anomalous data points in 550 spaceflight observations. The technique's utility has been demonstrated through the successful 551 552 identification of various physical phenomena in a variety of data products from several missions, and its potential applicability to low-cost spaceflight hardware has been discussed. Additionally, 553 several avenues for potential future research utilizing the proposed technique have been 554 555 highlighted.

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566 **Open Research**

The CASSIOPE/Swarm-Echo MGF data used in this manuscript is publicly available at 567 https://epop-data.phys.ucalgary.ca/. The MMS FGM, FPI, and SITL data used in this manuscript 568 is publicly available through https://lasp.colorado.edu/mms/sdc/public/. Code used to implement 569 the algorithm described in this manuscript, along with sample data used to illustrate the technique's 570 performance. currently stored 571 are at https://drive.google.com/drive/folders/1151OzBKW9Mgf6xVQ8P_bGLMmxTfBjHyG?usp=sha 572 ring. Upon acceptance of this manuscript, the code and sample data will be moved to a University 573 574 of Maryland institutional repository, or similar digital repository, for long-term storage and reuse.

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Generalized Time-Series Analysis for In-Situ Spacecraft Observations:
 Anomaly Detection and Data Prioritization using
 Principal Components Analysis and Unsupervised Clustering

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12 Key Points:

- Spacecraft often cannot transmit all measurements to Earth at full cadence due to
 telemetry bandwidth limitations.
- Many missions must implement complex data prioritization schemes to ensure only the
 highest-priority data is transmitted at high cadence.
- The proposed data prioritization technique is highly generic, compatible with inexpensive
 hardware, and suitable for low-cost missions.
- 19

20 Abstract

In-situ spacecraft observations are critical to our study and understanding of the various 21 phenomena that couple mass, momentum, and energy throughout near-Earth space and beyond. 22 However, on-orbit telemetry constraints can severely limit the capability of spacecraft to transmit 23 high-cadence data, and missions are often only able to telemeter a small percentage of their 24 25 captured data at full rate. This presents a programmatic need to prioritize intervals with the highest probability of enabling the mission's science goals. Larger missions such as the Magnetospheric 26 Multiscale mission (MMS) aim to solve this problem with a Scientist-In-The-Loop (SITL), where 27 a domain expert flags intervals of time with potentially interesting data for high-cadence data 28 downlink and subsequent study. Although suitable for some missions, the SITL solution is not 29 always feasible, especially for low-cost missions such as CubeSats and NanoSats. This manuscript 30 presents a generalizable method for the detection of anomalous data points in spacecraft 31 observations, enabling rapid data prioritization without substantial computational overhead or the 32 need for additional infrastructure on the ground. Specifically, Principal Components Analysis and 33 One-Class Support Vector Machines are used to generate an alternative representation of the data 34 and provide an indication, for each point, of the data's potential for scientific utility. The 35 technique's performance and generalizability is demonstrated through application to intervals of 36 observations, including magnetic field data and plasma moments, from the CASSIOPE e-37 38 POP/Swarm-Echo and MMS missions.

39 Plain Language Summary

Measurements captured by spacecraft are necessary to our understanding the space environment 40 near Earth and throughout our solar system. However, spacecraft can often only transmit a small 41 portion of the data they capture back to Earth. This means that many spacecraft must prioritize 42 intervals of data that have the highest probability of helping to further our understanding of these 43 environments. Some missions utilize humans, on Earth, to help select these scientifically important 44 intervals. This solution, called the Scientist-In-The-Loop, can be too expensive or 45 programmatically complex for many small missions to implement. This manuscript presents a 46 technique for the detection of anomalous events in spaceflight measurements using statistical 47 analysis and machine learning. These detected anomalies can be used to prioritize data that has a 48 high probability of scientific relevance. Further, the proposed technique is highly generalizable 49 and computationally lightweight, making it suitable for a variety of missions. Several case studies 50 from multiple existing missions will be analyzed throughout this paper. 51

52 **1 Introduction**

Magnetic field sensors are one of the many science instruments that have been a 53 fundamental part of space exploration since its inception. Some of the first satellites, such as the 54 late-1950's Sputnik 3 and Explorer 6, carried fluxgate magnetometers to collect scientific data 55 (Gordon & Brown, 1972). Since then, the science of magnetometry and spaceflight has been 56 57 advanced such that we can make magnetic field measurements of far-flung bodies such as asteroids (Weiss et al., 2023), Mars (Connerney et al., 2015), Jupiter (Connerney et al., 2017), and the Sun 58 (Bale et al., 2016). The need to understand fundamental physical processes in space, such as 59 magnetic reconnection, has driven requirements for the telemetry of measurements at higher and 60 higher cadences (Phan et al., 2016). Additionally, our desire to enable comprehensive 61 understanding, nowcasting, and forecasting of Earth's near-space environment has led to the 62

development of the inexpensive CubeSat form factor and a greater number of multi-spacecraft missions. CubeSats allow cost effective proliferation of measurements that can vastly improve our models through data assimilation and machine learning techniques. The last several years have seen the emergence of large constellations, leading to a multitude of challenges, particularly in the handling and telemetry of the massive quantities of raw data available onboard each spacecraft (Liddle et al., 2020; Zhan et al., 2020).

Many missions, due to programmatic constraints on telemetry rates, are unfortunately 69 unable to downlink all of their captured data to Earth for analysis. Instead, mission operators and 70 science teams must make decisions about which intervals of time to transmit high-cadence data 71 (i.e., burst data). Some missions will transmit a lower-cadence data product (i.e., survey data) 72 during intervals deemed less important (Lepping et al., 1995), and some missions will simply not 73 telemeter these intervals (Yau & James, 2015). Intervals of burst data to be telemetered are 74 typically determined by estimated spacecraft position, by humans on the ground diligently 75 monitoring low-cadence data, or by carefully calibrated onboard algorithms which search the high-76 cadence data for mission-specific triggers. 77

The Magnetospheric Multiscale (MMS) mission (Burch et al., 2016) utilizes a particularly 78 thorough approach to the identification and prioritization of burst data to telemeter to the ground 79 (Baker et al., 2016). During spacecraft traversal through predetermined regions of interest, the 80 MMS instruments always capture data at their high cadence burst rates. This high-cadence data is 81 stored onboard while the lower-cadence survey data is telemetered to the ground and analyzed. If 82 83 the survey data shows potentially interesting phenomena, short intervals of burst data can be downlinked from the spacecraft. The MMS mission prioritizes burst data using two techniques: 84 85 the Scientist in the Loop (SITL) and the Automated Burst System (ABS).

The ABS, as its name indicates, runs automatically onboard each spacecraft and provides 86 a data ranking metric to be downlinked alongside the survey data (Baker et al., 2016). This system 87 uses data quality indicators calculated by each instrument to rank the available burst data in a 88 downlink prioritization queue. The last item in the queue will be the first to be overwritten should 89 a higher-ranking interval be identified. Although 34 data quality indicators are available for burst 90 91 triggering, the early mission used only large gradients in the Z_{GSM}-component of the measured magnetic field to prioritize burst intervals while the data quality indicators were characterized. 92 After two years of careful parameterization, ~6 data quality indicators are now used by the ABS 93 94 for prioritization of burst data containing mission-specific phenomena of interest (Argall et al., 2020). 95

The SITL is a manual option that can validate or override the selections made by the ABS. A domain expert – with access to MMS survey data, spacecraft-calculated trigger metadata, and data from other satellites or ground systems – determines the priority of data for downlink using specialized software, ensuring that data with high scientific significance is telemetered (Baker et al., 2016).

Both the ABS and SITL burst selection schemes for data prioritization require substantial scientific infrastructure and potentially costly overhead in their implementation. Extensive onboard triggering logic, years of parameter characterization, dedicated interval labeling time from experts during in-situ mission operations, and bespoke burst prioritization software are almost certainly infeasible for low-budget missions. One method has been recently proposed which aims to reduce the need for infrastructureintensive SITL activities on MMS using supervised machine learning (Argall et al., 2020). Although their technique shows excellent results and enables a great reduction in the reliance on manual labeling tasks, it requires a large set of expert-labeled data during the training of their segmentation network and therefore does not eliminate the need for the SITL infrastructure entirely.

This manuscript proposes the use of a common dimensionality reduction technique, 112 coupled with unsupervised clustering, to provide a robust and generalizable method for detecting 113 anomalous intervals of time series spacecraft observations. This method is intended for use as a 114 component of a drop-in burst data prioritization system for missions where the infrastructure and 115 cost associated with more sophisticated and mission-specific solutions are not feasible. Although 116 the SpaceX Starlink constellation currently dominates the Low Earth Orbit environment with over 117 2,000 satellites currently in orbit and approval granted for 12,000 total satellites, it is unlikely to 118 remain the only major constellation in orbit (Ma et al., 2023; McDowell, 2020). These mega-119 constellations are a heavy burden on ground systems, requiring complex protocols for dealing with 120 telemetry, command, and tracking (Berner, 2019). The proposed tool would be an invaluable asset 121 for such constellation missions, enabling a higher degree of distributed autonomy in their space 122 operations. 123

The following sections of this manuscript describe the proposed technique and demonstrate its performance with several case studies on observational measurements obtained from the CASSIOPE e-POP/Swarm-Echo spacecraft and one of the Magnetospheric Multiscale mission satellites.

128 2 Methodology

129 2.1 Dimensionality Reduction via Principal Components Analysis

130 Principal Components Analysis (PCA) is one of the oldest and most popular multivariate

statistical analysis techniques used to reduce the dimensionality of large datasets (Jolliffe &

132 Cadima, 2016). Mathematically, PCA is performed by identifying the eigenvectors of the

133 covariance matrix associated with the data matrix under observation (X) via

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$\boldsymbol{X}^{T}\boldsymbol{X} = \boldsymbol{V}\boldsymbol{\Lambda}\boldsymbol{V}^{T} \tag{1}$

where the columns of *V* correspond to the eigenvectors and the diagonal elements of Λ are the associated eigenvalues. For convenience, let each eigenvector V_i be ordered by the magnitude of its associated eigenvalue.

138 The projection and subsequent dimensionality reduction can be realized through

$$\boldsymbol{P} = \boldsymbol{X} \boldsymbol{V}_{\boldsymbol{q}} \tag{2}$$

where V_q is a matrix whose columns are only the first q eigenvectors from V. Throughout this manuscript, the dimensionality of the output projection is fixed to two (i.e., q = 2) in order to reduce the computational complexity associated with the analysis of the projected data.

143 The specific data matrix being analyzed in this manuscript is generated by concatenating 144 *R* consecutive time intervals, of length *L*, from the original time series (x, with length *N*) into a 145 single matrix via 146

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$$\mathbf{X} = \begin{bmatrix} x(1) & x(L+1) & \dots & x(RL-L+1) \\ \dots & \dots & \dots & \dots \\ x(L) & x(2L) & \dots & x(RL) \end{bmatrix}.$$
 (3)

147 It should be noted that X is a reduced version of the trajectory matrix passed into PCA by the 148 Singular Spectrum Analysis (SSA) technique (Finley et al., 2023). Both SSA's trajectory matrix 149 and the data matrix used throughout this paper are constructed to enable information about the 150 temporal variation of a signal to be utilized. The reduced trajectory matrix (i.e., no overlapping 151 samples in each consecutive column) was used here to reduce the computational complexity 152 associated with the overall algorithm.

An example of this variation on PCA, applied to one axis of the vector magnetometer data captured by the CASSIOPE e-POP/Swarm-Echo magnetic field instrument (Wallis et al., 2015; Yau & James, 2015), is shown in Fig. 1. Figure 1(a) illustrates the 2.5-minute interval of data, at a sampling rate of 160 Hz, to be processed. Figure 2(b) shows the two-dimensional representation, given by P, of the data following the application of PCA (with q = 2) on a data matrix constructed from 0.5-second consecutive intervals taken from the signal in Fig. 1(a).



Figure 1: Dimensionality reduction via application of PCA on concatenated 0.5-second intervals of magnetic field data. (a)
 Inboard magnetometer data from CASSIOPE e-POP/Swarm-Echo MGF; (b) Two-dimensional representation of the 0.5-second
 intervals of (a) following PCA while retaining only two principal components.

Each point in Fig. 1(b) is a projection of one of the 0.5-second intervals of Fig. 1(a). It can 163 be seen that many of these points are clustered tightly near the origin, whereas some of the points 164 are outlying near the periphery. This implies that the majority of the time intervals exhibit similar 165 behavior when represented using only the first two principal components (i.e., those that describe 166 167 the largest variance in the original data matrix). However, some intervals show very different behavior in terms of these maximum-variance components. Automatic clustering of this two-168 dimensional representation should reveal anomalous time intervals in the original signal and is 169 discussed in detail in Sec. 2.2. 170

171 2.2 Clustering with One-Class Support Vector Machine

Machine learning techniques have become increasingly popular in the various space 172 physics research domains. Successful application of these techniques has been seen in methods for 173 auroral image classification (Clausen & Nickisch, 2018), recreating magnetohydrodynamic 174 environments from sparse sample spaces (Bard & Dorelli, 2021), space weather forecasting 175 (Camporeale, 2019), and many others. However, machine learning models can often be complex, 176 requiring large quantities of training data and computational resources. Once trained, these large 177 and complex models are often treated as 'black boxes,' and can lack interpretability (Angelov et 178 al., 2021). To increase potential applicability to low-cost and in-situ spaceflight hardware, a 179

machine learning-based clustering solution that is computationally efficient and easily understoodmust be utilized instead of a more complex model.

Support Vector Machines (SVMs) are a popular means of performing classification tasks 182 throughout a variety of fields including the biomedical sciences (Zhou et al., 2005) and industrial 183 engineering (Shin et al., 2005). This data labeling technique has seen widespread adoption due to 184 its high degree of robustness and interpretability (Hearst et al., 1998). Traditional SVMs are trained 185 by first projecting the labeled training data to a higher dimension feature space using a user-186 selected *kernel*. Next, a hyperplane that best separates the classes is calculated, although a slack 187 parameter is considered in this optimization. This slack parameter enables the trained SVM to 188 handle a small number of data points that cannot be separated using a hyperplane in the higher-189 190 dimension feature space, which is a common situation in real-world datasets (Noble, 2006). This trained SVM can now be used to classify new data not seen during the training process. 191

Slight modification of the traditional SVM framework leads to a technique known as the 192 One-Class Support Vector Machine (OC-SVM), a common unsupervised approach to data 193 classification and anomaly detection (Yin et al., 2014). These OC-SVMs operate in a similar 194 fashion to the traditional SVM but calculate a hyperplane that optimally separates the data from 195 the origin, not by separating pre-labeled classes (Amer et al., 2013). Here, the primary user-defined 196 control parameter is v, which lies in the range (0,1] and determines the upper bound on the number 197 of allowed errors and a lower bound on the number of data points used when calculating the 198 199 separating hyperplane (Chang & Lin, 2001).



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201 Figure 2: One-Class Support Vector Machine clustering applied to the reduced-dimension data illustrated in Fig. 1.

Figure 2 illustrates the result of passing the two-dimensional output of PCA, shown in Fig. 1(b), through an OC-SVM. This OC-SVM was trained with a Gaussian kernel and a v-value of 0.3. Points shown in red were those that were considered anomalous, whereas the points shown in blue were considered nominal. Since each point in Fig. 2 represents a 0.5-second interval of the original input shown in Fig. 1(a), the associated labels can be directly applied to each interval in the original input time series. The result of this inversion procedure, and additional examples, will be discussed in detail in Sec. 4.

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210 3 Data and Preprocessing

211 3.1 CASSIOPE e-POP/Swarm-Echo MGF

One primary source of the data analyzed in previous sections and throughout the remainder 212 of this manuscript is the CASSIOPE/Swarm-Echo Magnetic Field instrument (Wallis et al., 2015; 213 Yau & James, 2015). The Magnetic Field instrument (MGF) consists of a pair of identical fluxgate 214 magnetometers mounted inline on a single boom at approximately 0.6 m and 0.9 m from the body 215 216 of the spacecraft. Both magnetometers capture the local magnetic field at a rate of 160 Hz and downlink the data when telemetry constraints allow. Although the magnetometer pair (i.e., 217 gradiometer) could be used to mitigate local interference from the host spacecraft and improve 218 data fidelity (Finley et al., 2023), the measurements used in this manuscript were taken from only 219 the inboard magnetometer mounted closer to the spacecraft. This provides some insight into how 220 the proposed anomaly detection technique handles data contaminated by local interference, which 221 222 is almost constantly observed at the magnetometers as high-frequency oscillations caused primarily by the spacecraft's attitude control systems (i.e., reaction wheels). The magnetometer 223 measurements used in this manuscript have had a near-DC baseline removed using a 20-s moving 224 average prior to analysis and visualization. 225

226 3.2 Magnetospheric Multiscale Mission FGM

Another source of magnetic field data used in the remaining sections of this manuscript is 227 the Magnetospheric Multiscale mission (MMS) Magnetometers (Burch et al., 2016; Russell et al., 228 2016). Although data is available from all four MMS satellites, only data from MMS1 was utilized 229 throughout this manuscript. The MMS Magnetometers consist of a near-identical pair of fluxgate 230 sensors, with each sensor mounted at the end of two separate 5-meter booms. The measured data 231 is reported as a high-fidelity triaxial vector data product (called FGM) with three possible sampling 232 rates: slow survey at 8 Hz; fast survey at 16 Hz; and, burst data at 128 Hz. In this manuscript, only 233 fast survey and slow survey data were utilized. If both fast and slow survey data were present in 234 the time interval to be analyzed, the data was resampled to match the slow survey data rate for 235 consistency of analysis. As with the CASSIOPE data discussed in Sec. 3.1., a 20-s moving average 236 has been removed from the MMS data prior to analysis and visualization. 237

238 3.3 Magnetospheric Multiscale Mission FPI

An additional set of data used in the remaining sections of this manuscript is the MMS Fast 239 240 Plasma Investigation (Burch et al., 2016; Pollock et al., 2016). As with the magnetic field data, only data from the MMS1 satellite was used in this manuscript. The Fast Plasma Investigation 241 (FPI) for MMS comprises multiple top-hat electrostatic analyzers (Carlson et al., 1982) to 242 determine in situ the fluxes of electrons and ions as functions of energy and direction. The FPI, its 243 measurements, and methods of computation are described in detail in (Pollock et al., 2016). FPI 244 acquires a full 3D set of electron samples (32 energies \times 32 azimuths \times 16 polar sections) every 245 30 ms, and an equivalent set of ion samples every 150 ms. When telemetered to the ground, these 246 samples are employed to compute electron and ion fluid parameters such as number density, bulk-247 flow velocity, pressure, and others. Each parameter is computed as a summation of fluxes weighted 248 249 by appropriate physical factors. These parameters are essential quantities required for many scientific analyses of space plasmas, for example in comparisons of observations with the output 250 of magnetohydrodynamic (MHD) simulations that predict the overall behavior of plasma as a fluid. 251

The ion parameters (i.e., number density and velocity) shown in this manuscript are reported at the Fast Survey rate of ~0.22 Hz (i.e., captured at full cadence and averaged once every 4.5 s).

For the FPI, approximate fluid-like parameters are also computed in a simplified fashion by the instrument processor onboard the spacecraft. The computation is approximate because the summations are not weighted by the proper factors necessary to obtain true physical quantities. However, it has been shown that these onboard quantities, known as pseudo-moments, can be rescaled to serve as proxies for the true physical parameters (Barrie et al., 2018). As a result, analysis of the derived plasma moments using the proposed anomaly detection technique is indicative of the technique's performance when applied to the onboard pseudo-moments.

261 3.3 Magnetospheric Multiscale Mission SITL

To verify the capability of the proposed technique to identify intervals of time relevant to 262 the MMS mission's science objectives, this manuscript also includes data from the MMS Scientist-263 in-the-Loop (SITL) report. MMS is the first mission with both the spatial and temporal resolution 264 to resolve electron-scale dynamics. This requires MMS to capture much more data than it can 265 telemeter to the ground. As discussed in Sec. 1, this has led to the development of MMS' burst 266 management system, consisting of the Automated Burst System (ABS), Scientist-in-the-Loop 267 (SITL), and the Ground Loop System (GLS). The SITL, specifically, is a role passed among MMS 268 team member volunteers who search through the survey data to identify and select time intervals 269 that may contain relevant events for burst-mode downlink. Survey data has insufficient resolution 270 to capture electron-scale dynamics, so the SITL must over-select events to ensure that mission-271 272 critical science data is captured. The SITL is guided by mission-level science objectives that enable assignment of a figure of merit (FOM) value to each interval examined (Argall et al., 2020; 273 Hasegawa et al., 2023; Phan et al., 2016). These FOMs subsequently inform the mission of the 274 highest-priority data to be selected and downlinked at burst rate. Each SITL selection is 275 accompanied by a short description that is searchable and parsable. These descriptions have 276 previously been used to train a supervised machine learning model that uses the SITL report to 277 make future predictions about which data should be selected for downlink (Argall et al., 2020). 278 This model is installed in the near real-time data processing GLS so that predictions can be made 279 280 as soon as the preliminary low-cadence data is downlinked in order to guide the SITL (along with the ABS selections). The SITL data itself is not publicly available, as it is not considered science-281 quality, but the reports describing the selected events and their time range can be searched through 282 the MMS Mission Events webpage (https://lasp.colorado.edu/mms/sdc/public/about/events/#/). 283 Additional tools, such as PyMMS, have been developed that enable rapid searching of these reports 284 (Argall et al., 2022). 285

286 **4 Results**

287 4.1 Observations of Current Sheets with Embedded Alfvén Waves (CASSIOPE)

288 Section 2 illustrated (through Fig. 1 and Fig. 2, respectively) the dimensionality reduction 289 and unsupervised clustering techniques utilized in the proposed method of automated anomaly 290 detection by applying them to a short interval of magnetic field data captured by the CASSIOPE 291 e-POP/Swarm-Echo inboard magnetometer. Figure 3(a) shows the proposed method's output after 292 inverting the labels given to each point shown in Fig. 2 back onto the original magnetometer data 293 shown in Fig. 1(a), as described in Sec. 2.2. Points in blue are those considered nominal by the 294 technique, whereas points labeled red have been flagged as anomalous or potentially scientifically

relevant. Figure 3(b) provides the spectral content for the interval shown in Fig. 3(a). The spectral 295 content and time series for this 2.5-minute interval shows previously identified Alfvénic activity 296 embedded in a large current sheet at roughly 6:48:55 UTC (Miles et al., 2018), with a smaller 297 current sheet observed near 6:48:10 UTC. The current sheet containing embedded Alfvénic 298 activity is highlighted in gray in Fig. 3(a). It can be seen that the proposed method of anomaly 299 detection was able to accurately identify these interesting intervals containing broadband magnetic 300 activity, even in the presence of the potentially obfuscating reaction wheel interference observed 301 at ~15 Hz. 302





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308 4.2 Observations of Spacecraft Maneuvers (CASSIOPE)

The next interval of data analyzed in this section is a 20-minute interval of CASSIOPE e-309 POP/Swarm-Echo magnetometer data captured when the spacecraft was performing maneuvers. 310 As a result, the spacecraft's four reaction wheels were changing their spin frequency rapidly during 311 this period. Figure 4(a) shows the labeled time series output by the method with a window length 312 of five seconds. Anomalous intervals are shown in red, normal points are shown in blue. Figure 313 314 4(b) shows the spectral content of the interval shown in Fig. 4(a). Highly dynamic high-frequency activity resulting from the reaction wheels' diverging operational rates can be seen for a five-315 minute interval starting at approximately 4:55:00 UTC, with other low-frequency perturbations 316 occurring throughout the total 20-minute interval. It can be seen that the detected anomalies 317 directly correspond to changes in behavior in the measured magnetic field due to the spacecraft 318 maneuvers. This example helps to both illustrate the ability of the proposed technique to 319 simultaneously identify dynamic behavior in both high- and low-frequency bands, as well as 320 identify intervals critical to spacecraft operations (i.e., spacecraft maneuvers). 321



Figure 4: Proposed method of anomaly detection applied to interval of CASSIOPE e-POP/Swarm Echo MGF data containing
 highly dynamic high-frequency signatures caused by spacecraft reaction wheels. (a) Time series with anomalous intervals plotted
 in red, nominal in blue. (b) Spectrogram associated with the data in (a).

327 4.3 Observations of Interplanetary Shocks and EMIC Waves (MMS)

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The next period of data analyzed in this manuscript is a full 24 hours of magnetic field data 328 captured by the MMS magnetometer suite in December 2015. Figure 5(a) shows the result of the 329 proposed method of anomaly detection applied to the full day of data with a window length (L) of 330 5 minutes. As in the previous examples, points shown in red correspond to detected anomalies and 331 332 points shown in blue have been identified as nominal. Figure 5(b) shows the spectrogram associated with the magnetic field measurements in Fig. 5(a). Note that this data has been 333 resampled to match the slowest sampling rate present in the interval (i.e., 8 Hz), as described in 334 Sec. 3.2. The frequent broadband magnetic phenomena seen throughout the first ~40% of this 24-335 hour interval is attributed, by the MMS SITL, to observations of the bow shock and magnetopause. 336 The broadband magnetic activity seen in the last ~15% of the day was similarly reported by SITL 337 to correspond to observations of the magnetopause. The large-amplitude (i.e., > 200 nT) 338 phenomena at approximately 17:00 UTC corresponds to the perigee of the MMS1 spacecraft. This 339 day of data, which was analyzed extensively by Engebretson et al. (2018), also contains MMS' 340 observations prior to, during, and after an interplanetary shock. The resultant compression of the 341 magnetosheath was observed by MMS at approximately 13:24 UTC, with structured EMIC wave 342 activity occurring before and after. The gray highlighted region in Fig. 5(a) corresponds to the 343 region of interest (from 13:00-14:00 UTC) containing the majority of this activity (Engebretson et 344 al., 2018). It can be seen that the proposed method of anomaly detection is able to accurately label 345 a significant portion of the interesting magnetic field data occurring at bow shock and 346 magnetopause crossings before noon and near midnight. In addition, the technique also labels 347 several of the large-amplitude perturbations in the magnetic field measurement near and during 348 the 13:00-14:00 UTC region of interest. Figures 6(a) and 6(b) provide a zoomed view of this 349

region's time series and spectral content, respectively. No change has been made to the anomaly 350 labeling, which still corresponds to the result when the proposed method is applied to the full 24-351 hour period. It is important to note that although the proposed technique has successfully identified 352 the broadband signal corresponding to the magnetosheath compression, much of the structured 353 EMIC wave activity occurring during this region of interest has not been labeled as anomalous. 354 This result is obtained due to the input parameters selected for this example. The next example, 355 shown in Fig. 7, will illustrate the effect of varying the input parameters when the anomaly 356 detection technique is applied. 357



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Figure 5: Anomaly detection technique applied to a 24-hour interval of MMS FGM data containing a variety of scientifically
 interesting phenomena including observations of the bow shock, magnetopause, and compressions of the magnetopause due to an
 interplanetary shock. (a) Time series FGM measurements with anomalous intervals identified by the proposed technique plotted
 in red, nominal intervals plotted in blue. (b) Spectrogram associated with the data shown in (a).



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Figure 7 illustrates the impact of changing the signal length (N) and window length (L) on 368 the performance of this technique. Specifically, Figure 7(a) shows the labeling output by the 369 proposed method of anomaly detection when applied to only the 13:00-14:00 UTC region of 370 interest highlighted in Fig. 5(a) and shown in Fig. 6(a). Note that this impacts the signal length, N, 371 without changing the window length, L. Although more of the structured wave activity in this 372 region is correctly labeled as anomalous, this result may still not be sufficient for some 373 applications. Reducing the window length (L) from five minutes to one second, as seen in Fig. 374 7(b), provides a more detailed labeling of the anomalous samples within the total interval. As a 375 result, much more of the structured EMIC wave activity occurring after the magnetosheath 376 incursion is identified as anomalous. Although the parameter space inherent to the proposed 377 method of anomaly detection is small, these results clearly show the relevance of the input 378 parameters to the method's result. Selection of these parameters, and their effect on the method's 379 computational complexity, will be discussed in greater detail in Sec. 5.1. 380



Figure 7: Illustration of the impact of the user-defined input parameters on the output of the proposed anomaly detection, shown for the same data seen in Fig. 6. (a) Time series output by the anomaly detection technique for a signal length of one hour and a window length of five minutes. (b) Time series output by the anomaly detection technique for a signal length of one hour and a window length of one second.

386 4.4 Observations of Magnetopause Crossings and the Bow Shock (MMS)

The next experiment in this manuscript demonstrates the applicability of the proposed 387 anomaly detection technique to spacecraft observations other than magnetic field measurements. 388 Specifically, the technique is applied to a large interval of magnetic field data, as well as ion 389 number density and ion velocity, as measured by the MMS spacecraft on 15 May 2023. Figure 390 8(a) shows a 24-hour period of magnetic field measurements, Fig. 8(b) and Fig. 8(c) show the ion 391 number density and ion velocity corresponding to the same period. The points illustrated in red are 392 those considered anomalous by the proposed technique when it was applied to the total interval, 393 whereas points shown in blue are considered nominal. It should be noted that the last 3.75 hours 394 of ion measurements were not available because the spacecraft transitioned from Fast Survey to 395 Slow Survey rate, and the FPI does not operate during Slow Survey mode. This period of missing 396 data has been padded with nominal-labeled zeroes, after the anomaly detection algorithm was 397 applied, for consistency with the magnetic field data. The gray region highlighted in Fig. 8 398 corresponds to a region of data containing various physical phenomena identified by the MMS 399 SITL, such as observations of the bow shock, magnetopause, and boundary layer crossings. These 400 SITL identifications, including the time range and the associated Figure of Merit (FOM) are 401 detailed in Tab. 1. A higher FOM is assigned to a selection that should be downlinked at higher 402 403 priority, with a maximum value of 199 being assigned to typical events, while higher FOMs are reserved for calibration intervals or other special events (Argall et al., 2020). 404

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408 Table 1: MMS SITL report for 15 May 2023. Columns 1-2 provide the start and stop time for each interval of interest. Column 3
 409 shows the Figure of Merit (FOM) associated with each interval. Column 4 shows SITL remarks for each interval.

Start Time (UTC)	Stop Time (UTC)	Figure of Merit (FOM)	Discussion
12:49:43	13:17:33	100	Partial Bow Shock
13:25:33	13:32:43	75	Partial Bow Shock
13:37:33	13:46:43	75	String of Partial Bow Shock Crossings
15:34:03	15:58:13	90	Partial Magnetopause
16:24:53	16:49:43	150	Full Medium Shear Magnetopause
17:51:43	17:56:13	55	Boundary Layer Traversal

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In addition to these SITL-identified phenomena, the magnetic field data shows frequent 411 broadband perturbations throughout the first half of the day, as well as a large data spike occurring 412 near midnight that corresponds to a perigee pass of the MMS satellite. It can be seen that, for all 413 three sets of input data, the anomaly detection technique successfully identifies much of the gray-414 highlighted region selected by the SITL as anomalous, but with a significant number of false 415 positives shown for the magnetic field data when compared to only the SITL selections. These 416 false positives in relationship to the SITL selection can be attributed to successful identification of 417 the broadband perturbations of the magnetic field during the first half of the day. This number of 418 false positives is greatly reduced in the labels associated with both the ion density and velocity, 419 implying that a combination of parameters classified by the proposed technique may be used in 420 the identification of only phenomena similar to those prioritized by the MMS SITL. 421



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Figure 9 provides a zoomed view of only the gray-highlighted region shown in Fig. 8. As 427 before, Fig. 9(a) - Fig. 9(c) show the magnetic field, the ion number density, and the ion velocity 428 429 measured by MMS. The nominal and anomalous labels assigned by the proposed anomaly detection technique correspond to the blue and red points, respectively. These labels are identical 430 to the labels seen in Fig. 8; only the time scale of the plot has been changed to visualize the 431 technique's output more clearly during the intervals selected by SITL. The regions highlighted in 432 gray in Fig. 9 correspond to the specific time periods identified by SITL (described in Tab. 1) as 433 containing observations of the bow shock, magnetopause, and boundary layer crossings. In a 434 similar trend to the larger period shown in Fig. 8, the magnetic field data contains a greater number 435 of false positive identifications of anomalous events when compared to only the SITL selections, 436 and the ion moments, or combinations of the ion moments and magnetic field, may provide a 437 greater degree of accuracy for this case study. 438

The example shown in Fig. 8 and Fig. 9 clearly demonstrates the generic applicability of 439 the anomaly detection algorithm to a variety of spaceflight data products. Additionally, it shows 440 that greater utility may be leveraged from the proposed method by analyzing its output when 441 442 applied to several data products. For example, a weighting scheme generated by some logical operation of the binary labels (nominal or anomalous) associated with the method's output, when 443 applied to multiple data products, may provide a reduced number of potential false positives 444 depending on the target application. This is left as a potential topic for future study, and Section 5 445 will discuss several other avenues for future work utilizing this technique. 446 447



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 449</sup> Figure 9: Zoomed view of the region of interest shown in Fig. 8., with SITL-identified observations of the bow shock,
 450 magnetopause, and boundary layer crossings. (a) FGM magnetic field data. (b) FPI ion number density. (c) FPI ion velocity.

451 **5 Discussion & Future Work**

452 5.1 Parameter Selection & Computational Complexity

This manuscript has demonstrated the capabilities of the proposed method of anomaly detection to analyze data from a variety of instruments and successfully identify various interesting phenomena; however, the example shown in Fig. 5 through Fig. 7 has illustrated the need for appropriate parameter selection to enable the highest possible scientific return. The proposed technique has a small parameter space, with only three variables impacting the output result. Thus, each of these parameters plays a critical role in the outcome of the analysis, as well is its computational complexity.

The input signal length (N) determines the total length of data to be analyzed when 460 determining anomalous activity. The window length, L, determines the length of the consecutive 461 observations used as features when performing PCA. Finally, the tuning parameter v used in the 462 OC-SVM is, practically, the fraction of the consecutive observations that can be considered 463 anomalous when the data is clustered. In practice, this means that the input signal length (N)464 must be large enough that the majority of the data (i.e., at least the fraction given by $1 - \nu$) 465 should be considered nominal, based on the target application. The window length must also be 466 carefully considered when applying the technique as it corresponds to the scale on which 467 anomalies are detected. If large, slowly varying changes are to be identified, a longer window 468 length might be utilized; if small, rapid changes must be flagged, a shorter window length may 469

470 be more suitable.

Changes in these parameters also have a significant impact on the computational 471 complexity of the previously described technique. PCA, which is one of the fundamental 472 building blocks for this algorithm, has two basic steps: computation of the covariance matrix, 473 which has a complexity of $O(L^2R)$ for a given window length of L and a given number of total 474 consecutive observations, R; and, eigendecomposition of the covariance matrix, which has a 475 complexity of up to $O(R^3)$ in the worst-case scenario (Zhang et al., 2015). For the OC-SVM, the 476 most computationally expensive component is the model training, with a complexity of up to 477 $O(R^3)$. However, this complexity has been proven to be reduced by approximately one order of 478 magnitude using more complex techniques such as sequential minimal optimization (Kang et al., 479 2019). 480

It can be seen that the window length, the number of consecutive observation intervals, and the OC-SVM tuning parameter have a direct impact on the computational complexity, or the detected anomalies associated with the proposed algorithm. As a result, users must tune these parameters as appropriate to suite the capabilities of their hardware and the data being processed.

485 5.2 Potential for Implementation on Low-Cost Hardware

Although PCA has historically been one of the most popular statistical analysis techniques 486 used to reduce the dimensionality of large datasets (Jolliffe & Cadima, 2016), until recently 487 relatively few studies have evaluated the potential for PCA's implementation on embedded 488 hardware (Korat & Alimohammad, 2019). Early efforts to perform PCA on embedded hardware 489 relied on bespoke Very Large Scale Integration (VLSI) integrated circuits with complex 490 architecture (Tung-Chien Chen et al., 2008), but more recent works have leveraged modern and 491 relatively inexpensive FPGA technologies to perform either portions of the PCA computations 492 (Ali et al., 2013) or complete implementations of the PCA algorithm (Bravo et al., 2010; Korat & 493 Alimohammad, 2019). Additional research has proven that embedded implementation of PCA is 494 generic and highly scalable, enabling substantial improvements in the computational speed of the 495 technique across a range of applications (Shahrouzi & Perera, 2019). 496

Support Vector Machines (SVMs), which fall into the broad category of machine learning algorithms, have historically provided excellent performance when classifying complex and continuous features (Saidi et al., 2021). Generally, machine learning techniques are considered computationally expensive and challenging to implement on embedded hardware (Sze et al., 2017). However, several recent studies have shown the potential for the SVM algorithm to be implemented on a variety of embedded devices including VLSI integrated circuits and FPGAs (Amezzane et al., 2020; Loukrakpam & Choudhury, 2020).

In addition to being suitable for implementation on embedded hardware, both Principal Components Analysis and/or Support Vector Machines have been previously utilized for spacecraft fault detection and diagnosis (Yu Gao et al., 2012), the onboard detection of anomalous behavior in CubeSat solar panels (Cespedes et al., 2022), and other intelligent decision making applications onboard spacecraft (Jallad & Mohammed, 2014). This illustrates that not only can PCA and SVM be implemented in hardware, but that they have history in successful implementation for spaceflight applications.

511 5.3 Semantic Labeling of Prioritized Data

512 This manuscript has proposed a technique for the automated binary classification of time-513 series data as either anomalous (i.e., potentially of high scientific importance) or nominal, one 514 useful avenue of future work would be to provide additional semantic labels for the high-priority

data. These labels could indicate whether an identified event falls into a particular class of 515 geophysical event, such as shocks, magnetopause crossings, or whistler-mode waves. Several 516 machine learning techniques have been previously developed for the identification of events in 517 spaceflight data archives (Fordin et al., 2023; Vech & Malaspina, 2021), although the 518 computational intensity or large dimensionality of some of these algorithms make them potentially 519 unsuitable for deployment on spaceflight hardware. The proposed anomaly detection technique 520 provides utility to the pursuit of semantic labeling in two ways: firstly, as a low-cost data reduction 521 tool it can reduce the number of samples that must be processed by more complex algorithms, 522 decreasing the overall time complexity of the problem; secondly, the binary-labeled data can serve 523 as a powerful input feature vector if data reduction is not desired or required, potentially increasing 524 525 model performance.

526 5.4 Generalizability

527 This manuscript has illustrated the applicability of the proposed method of anomaly 528 detection to magnetic field data from MMS and CASSIOPE/Swarm-Echo, as well as the ion 529 density and velocity moments from MMS. Although this shows promise for the generalizability 530 of the technique to other platforms and other physical observations, thorough exploration of this 531 generalizability to other spacecraft observations (such as ion and electron pressures and 532 temperatures, as well as the housekeeping measurements critical to spacecraft operations) remains 533 an avenue for future work.

Testing the generalizability of the clustering model trained on a specific interval of time against different intervals is also an interesting future project. If the clustering learned by an OC-SVM trained on one interval of time (e.g., one of the 24-hour periods of MMS data) is applied to subsequent intervals (e.g., the next several days of MMS data) with meaningful results, the computational complexity of the proposed technique would decrease by eliminating the need to retrain a model for every interval under observation, enabling more rapid in-situ application.

540 6 Conclusions

The scientific measurements captured by in-situ spacecraft are critical to our study of the 541 physical phenomena that control the flow of mass, momentum, and energy throughout our solar 542 system. However, due to burdens on ground systems as a result of a rapid increase in the number 543 of active spaceflight missions and the ever-growing need for higher-cadence data, spacecraft are 544 often unable to transmit all of their data to Earth at full rate. As a result, missions must develop 545 techniques that enable prioritization of specific intervals with a high probability of importance to 546 their science goals. The techniques that have historically been used by large missions have 547 provided excellent results but come with high design and implementation costs, leaving them 548 potentially unsuitable for application on low-cost missions. This manuscript has proposed a 549 generic technique for the prioritization of data through the detection of anomalous data points in 550 spaceflight observations. The technique's utility has been demonstrated through the successful 551 552 identification of various physical phenomena in a variety of data products from several missions, and its potential applicability to low-cost spaceflight hardware has been discussed. Additionally, 553 several avenues for potential future research utilizing the proposed technique have been 554 555 highlighted.

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566 **Open Research**

The CASSIOPE/Swarm-Echo MGF data used in this manuscript is publicly available at 567 https://epop-data.phys.ucalgary.ca/. The MMS FGM, FPI, and SITL data used in this manuscript 568 is publicly available through https://lasp.colorado.edu/mms/sdc/public/. Code used to implement 569 the algorithm described in this manuscript, along with sample data used to illustrate the technique's 570 performance. currently stored 571 are at https://drive.google.com/drive/folders/1151OzBKW9Mgf6xVQ8P_bGLMmxTfBjHyG?usp=sha 572 ring. Upon acceptance of this manuscript, the code and sample data will be moved to a University 573 574 of Maryland institutional repository, or similar digital repository, for long-term storage and reuse.

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