Why are ELEvoHI CME arrival predictions different if based on STEREO-A or STEREO-B heliospheric imager observations?

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Abstract

Accurate forecasting of the arrival time and arrival speed of coronal mass ejections (CMEs) is a unsolved problem in space weather research. In this study, a comparison of the predicted arrival times and speeds for each CME based, independently, on the inputs from the two STEREO vantage points is carried out. We perform hindcasts using ELlipse Evolution model based on Heliospheric Imager observations (ELEvoHI) ensemble modelling. An estimate of the ambient solar wind conditions is obtained by the Wang-Sheeley-Arge/Heliospheric Upwind eXtrapolation (WSA/HUX) model combination that serves as input to ELEvoHI. We carefully select 12 CMEs between February 2010 and July 2012 that show clear signatures in both STEREO-A and STEREO-B HI time-elongation maps, that propagate close to the ecliptic plane, and that have corresponding in situ signatures at Earth. We find a mean arrival time difference of 6.5 hrs between predictions from the two different viewpoints, which can reach up to 9.5 hrs for individual CMEs, while the mean arrival speed difference is 63 km s\$^{-1}\$. An ambient solar wind with a large speed variance leads to larger differences in the STEREO-A and STEREO-B CME arrival time predictions (\$cc = 0.92\$). Additionally, we compare the predicted arrivals, from both spacecraft, to the actual in situ arrivals at Earth and find a mean absolute error of 7.5 \$\pm\$ 9.5 hrs for the arrival time and 87 \$\pm\$ 111 km s\$^{-1}\$ for the arrival speed. There is no tendency for one spacecraft to provide more accurate arrival predictions than the other.

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Key Points:

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15	•	A comparison of CME arrival time and speed predictions from two vantage
16		points was carried out using ELEvoHI
17	•	A highly structured ambient solar wind flow leads to larger arrival time differ-
18		ences between STA and STB predictions

The assumption of a rigid CME front in ELEvoHI and other HI-based methods
 is most probably too simplistic

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21 Abstract

Accurate forecasting of the arrival time and arrival speed of coronal mass ejections 22 (CMEs) is a unsolved problem in space weather research. In this study, a comparison 23 of the predicted arrival times and speeds for each CME based, independently, on the 24 inputs from the two STEREO vantage points is carried out. We perform hindcasts 25 using ELlipse Evolution model based on Heliospheric Imager observations (ELEvoHI) 26 ensemble modelling. An estimate of the ambient solar wind conditions is obtained 27 by the Wang-Sheeley-Arge/Heliospheric Upwind eXtrapolation (WSA/HUX) model 28 combination that serves as input to ELEvoHI. We carefully select 12 CMEs between 29 February 2010 and July 2012 that show clear signatures in both STEREO-A and 30 STEREO-B HI time-elongation maps, that propagate close to the ecliptic plane, and 31 that have corresponding in situ signatures at Earth. We find a mean arrival time 32 difference of 6.5 hrs between predictions from the two different viewpoints, which can 33 reach up to 9.5 hrs for individual CMEs, while the mean arrival speed difference is 34 63 km s^{-1} . An ambient solar wind with a large speed variance leads to larger differ-35 ences in the STEREO-A and STEREO-B CME arrival time predictions (cc = 0.92). 36 Additionally, we compare the predicted arrivals, from both spacecraft, to the actual 37 in situ arrivals at Earth and find a mean absolute error of 7.5 ± 9.5 hrs for the ar-38 rival time and 87 ± 111 km s⁻¹ for the arrival speed. There is no tendency for one 39 spacecraft to provide more accurate arrival predictions than the other. 40

41 **1 Introduction**

Understanding the dynamics of coronal mass ejections (CMEs) in the heliosphere 42 is a key aspect of space weather research. CMEs are huge clouds of energetic and 43 magnetized plasma (Hundhausen, Stanger, & Serbicki, 1994) erupting from the solar 44 corona that may reach speeds of up to 3000 km s⁻¹. When they hit Earth, CMEs 45 can produce strong geomagnetic storms (Gosling, Bame, McComas, & Phillips, 1990; 46 Kilpua, Jian, Li, Luhmann, & Russell, 2012; Richardson & Cane, 2012; Srivastava 47 & Venkatakrishnan, 2004) causing communication and navigation system problems, 48 damaging satellites and can even cause power outages (Cannon, 2013). The need 49 for accurate predictions of CMEs, both CME arrival time and speed, is becoming 50 increasingly important (Owens, Lockwood, & Barnard, 2020), because humankind, 51 more than ever, depends on advanced technology. 52

Shortly after their eruption, CMEs can be observed in coronagraph images. Two
 of the few space-borne coronagraphs in operation are the Large Angle and Spectrometric Coronagraph (LASCO) C2 and C3 on-board the Solar and Heliospheric Observatory
 (SoHO; Brueckner et al., 1995). SoHO is situated in a Lissajous orbit around Lagrange point 1 (L1), about 1.5 million km upstream of Earth in the Sun-Earth line.

The launch of the Solar Terrestrial Relations Observatory (STEREO: Kaiser et 58 al., 2008) twin-spacecraft mission in 2006 provided an unprecedented opportunity to 59 observe CMEs from off the Sun-Earth line. The two spacecraft orbit the Sun slightly 60 closer (STEREO Ahead; STA) and slightly further (STEREO Behind; STB) than 61 Earth, leading to a separation of each spacecraft by about 22° per year from Earth in 62 opposite directions. Both spacecraft are equipped with the In-situ Measurements of 63 Particles and CME Transients (IMPACT; Luhmann et al., 2008) instrument package to measure solar wind speed, density and magnetic field and additionally host a suite 65 of imagers, such as the COR1 and COR2 (Howard et al., 2008) coronagraphs and the 66 heliospheric imagers, HI1 and HI2 (Eyles et al., 2009). The wide-angle HI cameras 67 provide observations of the heliosphere that allow us to track a CME from close to the 68 Sun out to the orbit of Earth, particularly in the ecliptic plane. 69

CMEs are optically thin structures that expand rapidly, and decreasing density
 lowers the line-of-sight integrated intensity in white-light data. As a consequence,

the tracking of CME fronts and the interpretation of HI image data is difficult. Furthermore, the plane-of-sky assumption is not valid, and we must assume a certain
longitudinal extent of the CME frontal shape.

CMEs may be influenced by different phenomena in the heliosphere, e.g. magnetic 75 forces close to the Sun, high-speed solar wind streams, or by other CMEs (Kay & 76 Opher, 2015; Lugaz et al., 2012; Möstl et al., 2015). The ambient solar wind can also 77 affect the kinematic and morphological characteristics of CMEs (e.g. Gopalswamy et 78 al., 2000; Gosling et al., 1990; Manoharan et al., 2004). A CME originating at a speed 79 80 much faster than the ambient solar wind speed is likely to experience deceleration while slow CMEs may accelerate during their propagation (Manoharan & Mujiber 81 Rahman, 2011; Richardson & Cane, 2010). Hence, not only the propagation direction 82 but also the kinematics and shape of CMEs can be altered (e.g. Y. D. Liu et al., 83 2014; Rollett et al., 2014; Ruffenach et al., 2015; Savani, Owens, Rouillard, Forsyth, & 84 Davies, 2010; Zuccarello et al., 2012). By tracking CMEs far out in the heliosphere, we 85 get an understanding of their interaction with the ambient solar wind and co-rotating 86 interaction regions. 87

Over the last decades, a vast number of CME prediction models have been de-88 veloped. They include empirical models, e.g. Effective Acceleration Model (EAM; 89 Paouris & Mavromichalaki, 2017), which use relationships between observable param-90 eters and the transit time. There are also drag-based models, (e.g. DBM; Vršnak et 91 al. 2013, DBEM; Dumbović et al. 2018, ANTEATR; Kay, Mays, and Verbeke 2020), 92 that make use of physics-based equations and account for drag between the ambient 93 solar wind and the CME. Other models make use of HI images, which require tech-94 niques to convert the measured elongation into radial distance. For example, the fixed 95 phi fitting (FPF; Rouillard et al., 2008; Sheeley, Walters, Wang, & Howard, 1999) 96 technique considers a CME as a single point, propagating at a constant speed, and 97 provides an estimate of the constant direction of the CME propagation relative to the 98 observer from the apparent acceleration within a sequence of HI images. The har-99 monic mean fitting (HMF; Lugaz, 2010; Möstl et al., 2011) method is similar except 100 that it describes a CME as a circle that remains attached to the Sun-center. The 101 self-similar-expansion fitting (SSEF; Davies et al., 2012; Lugaz et al., 2010; Möstl & 102 Davies, 2013) technique describes a CME as a circle having an increasing radius as 103 it propagates away from the Sun in such a way that it maintains a constant angular 104 width. FPF and HMF are extremes of the SSEF technique with a half width of 0° 105 and 90°, respectively. More sophisticated models combine both the drag-based ap-106 proach and HI observations (e.g. DBM fitting; Žic, Vršnak, and Temmer 2015, Ellipse 107 Evolution model based on HI observations, ELEvoHI; Amerstorfer et al. 2018; Rollett 108 et al. 2016). Finally, numerical models, which are computational heavy, solve mag-109 netohydrodynamic (MHD) equations (e.g., ENLIL; Odstrcil et al. 2004, EUHFORIA; 110 Pomoell and Poedts 2018) simulating the ambient solar wind in the full heliosphere 111 based on synoptic photospheric magnetic-field maps. CMEs are then injected into 112 these models to provide predictions regarding the arrival time and arrival speed at 113 different locations in the heliosphere. 114

As noted above, ELEvoHI aims to predict the arrival time and arrival speed of CMEs. The model assumes an elliptical shape for the CME front and incorporates the drag exerted by the ambient solar wind. Also, different sources of ambient solar wind speed (e.g. provided by numerical models) can serve as input to ELEvoHI (Amerstorfer et al., 2020). In its latest version, the model can be used with STEREO-A HI beacon mode data to provide near real-time CME arrival predictions.

This study assesses ELEvoHI to evaluate arrival time and speed predictions of past CMEs using STEREO HI science-quality data. We perform ELEvoHI ensemble predictions for 12 CMEs, where each CME is modeled using input data from STA and STB, separately. In an idealized case, in which a CME with an elliptical front Table 1: List of selected CMEs. ID and Date correspond to the unique identifier and the time of the first appearance of the CME in HI1 imagery, from the HELCATS catalog, for STA and STB spacecraft. ICMECAT ID is the identifier of the interplanetary coronal mass ejection (ICME) from an updated version of the HELCATS ICMECAT (Möstl et al., 2017), ICMEdate is the start time of the detected ICME and v_{ICME} is the measured in situ arrival speed obtained from the HELCATS ICMECAT.

Nr.	ID STA	Date STA	ID STB	Date STB	ICMECAT ID	ICME date	VICME	
							$[\rm km \ s^{-1}]$	
1	HCME_A	2010-02-03	HCME_B	2010-02-03	ICME_Wind_	2010-02-07	406 ± 2	
1	20100203_01	14:49	20100203_01	20:49	NASA_20100207_01	$18:04^{b}$	40012	
	HCME_A	2010-03-19	HCME_B	2010-03-19	ICME_Wind_	2010-03-23	 202⊥12	
4	20100319_01	22:09	20100319_01	20:09	MOESTL_20100323_01	$22:29^{c}$	292112	
	HCME_A	2010-04-03	HCME_B	2010-04-03	$\overline{\text{ICME}}_{\overline{\text{Wind}}}$	2010-04-05		
3	20100403_01	12:09	20100403_01	12:09	NASA_20100405_01	$07:55^{a}$	134±18	
	HCME_A	2010-04-08	HCME_B	2010-04-08	ICME_Wind_	2010-04-11	420 17	
4	20100408_01	06:49	$20100408_{-}01$	07:29	NASA_20100411_01	$12:20^{a}$	432±17	
	HCME_A	2010-05-23	HCME_B	2010-05-24	ICME_Wind_	2010-05-28		
5	20100523_01	22:09	20100524_01	00:09	NASA_20100528_01	$01:52^{a}$	370 ± 10	
	HCME_A	2010-10-26	HCME_B	2010-10-26	ICME_Wind_	2010-10-30		
0	20101026_01	15:29	20101026_01	16:10	MOESTL_20101030_01	$09:15^{b}$	360±9	
	HCME_A	2011-01-30	HCME_B	2011-01-30	ICME_Wind_	2011-02-04		
(20110130_01	20:09	20110130_01	18:49	MOESTL_20110204_01	$01:50^{a}$	373 ± 9	
	HCME_A	2011-02-14	HCME_B	2011-02-14	ICME_Wind_	2011-02-18	400 05	
8	20110214_02	22:49	20110214_02	22:09	MOESTL_20110218_01	$00:48^{a}$	493 ± 25	
	HCME_A	2011-09-06	HCME_B	2011-09-07	ICME_Wind_	2011-09-09	417 00	
9	20110906_02	23:29	20110907_01	03:29	MOESTL_20110909_01	$11:46^{a}$	417 ± 20	
	HCME_A	2012-01-23	HCME_B	2012-01-23	ICME_Wind_	2012-01-24		
10	20120123_01	04:49	20120123_01	05:29	MOESTL_20120124_01	$14:36^{a}$	013 ± 30	
11	HCME_A	2012-06-14	HCME_B	2012-06-14	ICME_Wind_	2012-06-16	480 20	
	20120614_01	16:09	20120614_01	16:09	MOESTL_20120616_01	$19:34^{a}$	489 ± 29	
	HCME_A	2012-07-12	HCME_B	2012-07-12	ICME_Wind_	2012-07-14		
12	20120712_02	18:49	20120712_01	18:09	MOESTL_20120714_01	$17:38^{a}$	615 ± 37	
<u>n)</u>								

a) shock arrival time

^{b)} time of density enhancement ^{c)} time of the magnetic flux rope

propagates in an ambient solar wind that is constant in space and time, one would 125 expect to get similar results for the arrival time and arrival speed from the two different 126 vantage points. Instead of inferring the propagation directions of the events under 127 study from HI images (e.g. FPF, SSEF), as was done by Amerstorfer et al. (2018), 128 we make use of coronagraph images and perform Graduated Cylindrical Shell (GCS; 129 A. Thernisien, Vourlidas, & Howard, 2009; A. F. R. Thernisien, Howard, & Vourlidas, 130 2006) reconstruction for each CME based on multi-vantage point coronagraph data. 131 Additionally, we apply a combination of the Wang-Sheeley-Arge (Arge, Odstrcil, Pizzo, 132 & Mayer, 2003) and the Heliospheric Upwind eXtrapolation (Owens & Riley, 2017; 133 Riley & Lionello, 2011) model (WSA/HUX model combination; Reiss et al., 2019, 2020) 134 to get an estimate of the ambient solar wind conditions in the heliosphere through 135 which the CME propagates. With the additional information about the propagation 136 direction of the CME and the modeled ambient solar wind, ELEvoHI is more likely to 137 give better arrival time and arrival speed predictions. 138

In Section 2, we describe our data selection process, including the data products, and list all of the studied CMEs. Section 3 deals with the ELEvoHI setup and how the input data required by the model is obtained. In Section 4, we present our results and give reasons for the difference in the model predictions based on STA and STB input data. The discussion and further implementations of the model are included in Section 5.

¹⁴⁵ 2 Data Preparation

We select a period between February 2010 and July 2012 during which the STEREO spacecraft had a separation angle from Earth of about 65° to 120° respectively, from which we study 12 CMEs. The HELCATS HICAT CME catalog lists about 700 entries over this time range (Harrison et al., 2018). However, our list is constrained to 12 events, since the CMEs have to:

- be observed by HI on both STA and STB spacecraft (as listed in the HIJoinCAT;
 Barnes et al., 2020)
- ¹⁵³ 2) propagate close to the ecliptic plane,
 - 3) have a corresponding in situ signature at Earth,
 - 4) be able to be tracked unambiguously in time-elongation maps.

Table 1 contains the list of selected CMEs with their unique identifier and the 156 time of their first observation in HI1 images (according to the HELCATS catalog 157 Version 6). The interplanetary CME (ICME) times and speeds are taken from version 158 2.0 of the HELCATS ICMECAT catalog (Möstl et al., 2020, see also the links in the 159 data section). The ICMECAT assimilates ICME catalogs from different spacecraft 160 into one consistent list, and was first published in Möstl et al. (2017). The ICME 161 date as observed by the Wind spacecraft is defined by the shock arrival time, or, if 162 no shock is present, the start of a density enhancement in front of the magnetic flux 163 rope (MFR). If neither is observed, the ICME start time is taken as the start time 164 of the MFR. The corresponding ICME speed is the mean proton bulk speed of either 165 the sheath region, the density enhancement ahead of the MFR, or the speed of the 166 MFR itself. The spread in the speed over the given interval for each event is indicated 167 in Table 1 by a standard deviation. For Table 1, some times in the ICMECAT were 168 originally taken from the Wind ICME catalog (Nieves-Chinchilla et al., 2018), while 169 other events that were not present in the Wind catalog were added by Möstl et al. 170 (2020) to the HELCATS ICMECAT. 171

To run ELEvoHI, we make use of several data products. Most important are 172 images from HI onboard STEREO. The HI instrument on each STEREO spacecraft 173 consists of two white-light wide-angle imagers, HI1 and HI2. HI1 has a field-of-view 174 (FOV) extending from $4^{\circ} - 24^{\circ}$ elongation (angle from Sun center) in the ecliptic and 175 HI2 has an angular FOV extending from $18.8^{\circ} - 88.8^{\circ}$ elongation in the ecliptic. The 176 nominal cadence of the HI1 and HI2 science data is 40 minutes and 120 minutes, 177 respectively. The science image bin size is 70 arc sec for HI1 and 4 arc min for 178 HI2. For the additional input parameters to ELEvoHI, we developed the Ecliptic cut 179 Angles from GCS for ELEvoHI tool (EAGEL, see Section 3.1). EAGEL ideally uses 180 coronagraph images from STEREO COR1/COR2 and from LASCO C2/C3 onboard 181 SoHO, but images from at least two different viewpoints are required. The FOV of 182 COR1 ranges from 1.4 - 4 R_{\odot} and COR2, from 2 - 15 R_{\odot}, while C2 has a FOV of 183 $1.5-6~\mathrm{R}_{\odot}$ and C3, $3.7-30~\mathrm{R}_{\odot}$ (all quoted in the plane-of-sky). The cadence of the 184 coronagraph science images is about 15 minutes. 185

3 Methods

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3.1 EAGEL (Ecliptic cut Angles from GCS for ELEvoHI)

In this section, we present a newly developed Interactive Data Language (IDLTM) tool called EAGEL (Ecliptic cut Angles from GCS for ELEvoHI). EAGEL allows the user to determine the propagation direction, ϕ , and the half width, λ , within the ecliptic plane, based on GCS reconstruction of a CME. To perform GCS reconstruction, coronagraph images from at least two vantage points (STEREO and/or LASCO) are required. EAGEL provides the routines to download the required coronagraph



Figure 1: GCS reconstruction (left) and ecliptic cut of the wireframe (right) for event #5. a) Top row from left to right: STB/COR2, LASCO/C3, STA/COR2. Bottom row: same as top row but with the GCS wire frame overlaid. b) Ecliptic cut (black) of the GCS wire frame. Red and green lines show the boundaries selected by either EAGEL or the user. The yellow line defines the ecliptic propagation direction, ϕ , of the CME. The half angle, λ , is the angle between one boundary and ϕ . The blue arrow indicates the direction to Earth.

images, combines all the functions to perform GCS reconstruction, and produces a 194 cut in the ecliptic plane. Standard pre-processing of the images is implemented in 195 EAGEL to make the CME features clearly visible to the user, who can decide be-196 tween using background-subtracted, running-difference, and base-difference images. 197 The user can then perform GCS reconstruction using the IDL SolarSoft procedure 198 rtsccguicloud. The top row of Figure 1a shows the coronagraph images (from left 199 to right: STB/COR2, LASCO/C3, STA/COR2) for event #5. The bottom row ad-200 ditionally shows the GCS wire frame (green mesh). In its current version, ELEvoHI 201 is a 2D prediction model giving results only in the ecliptic plane. Therefore, EAGEL 202 calculates the ecliptic part of the GCS wire frame and selects the boundaries of the 203 ecliptic cut (see red and green line in Figure 1b). The boundaries are defined to be 204 the outermost points of each side of the ecliptic cut with respect to the apex direction 205 from GCS reconstruction. This gives λ and ϕ , where the latter is defined to be exactly 206 in between the two boundaries. A plot is shown to the user (Figure 1b) and, if needed, 207 the boundaries can be changed manually. Once the user approves the selection, λ and 208 ϕ relative to Earth and to the two STEREO spacecraft are stored and can be used by 209 ELEvoHI. 210

In Table 2, we list the time (Date) of the STEREO coronagraph images used 211 to get λ and ϕ for each event. EAGEL then selects the SoHO coronagraph images 212 closest in time to the quoted date. The table further contains the GCS parameters 213 (Lon, Lat, TA, AR, HA), λ , and the CME ecliptic propagation angle, ϕ , relative to 214 Earth (ϕ_{Earth}) and relative to the two STEREO spacecraft (ϕ_{STA} and ϕ_{STB}) obtained 215 from EAGEL. Lon is the longitude (here given in Stonyhurst coordinates) and Lat the 216 latitude of the apex of the idealized hollow croissant shaped model. The tilt angle (TA)217 defines the tilt of the croissant and the half angle (HA) represents the angle between 218 the center of the footpoints. The aspect ratio (AR) describes the spatial extent of the 219 croissant. 220

When comparing *Lon* (longitude from GCS reconstruction) and ϕ_{Earth} (longitude relative to Earth from the ecliptic cut), it can be seen that the propagation direction

Table 2: GCS parameter obtained from fitting the hollow croissant shape the STEREO and SoHO coronagraph images. *Date*: time set in EAGEL to perform the reconstruction, *Lon*: GCS longitude (Stonyhurst coordinates), *Lat*: GCS latitude, *TA*: GCS tilt angle, *AR*: GCS aspect ratio, *HA*: GCS half angle, λ : half angle of the CME from the ecliptic cut, ϕ_{Earth} , ϕ_{STA} , ϕ_{STB} : propagation direction based on the ecliptic cut with respect to Earth, STA, STB, respectively.

Nr.	Date	Lon	Lat	TA	AR	HA	λ	$\phi_{ m Earth}$	ϕ_{STA}	$\phi_{\rm STB}$
		[°]	[°]	[°]		[°]	[°]	[°]	[°]	[°]
1	2010-02-03 15:54	355	-17	-1	0.33	30	36	-4	67	68
2	2010-03-19 17:39	23	-12	-7	0.29	19	30	22	44	93
3	2010-04-03 12:39	7	-19	15	0.39	30	38	9	58	81
4	2010-04-08 06:39	1	-10	-20	0.28	30	31	-2	70	69
5	2010-05-23 20:39	6	2	-15	0.48	18	35	-6	65	76
6	2010-10-26 14:39	18	-35	-28	0.51	30	18	-11	95	69
7	2011-01-30 21:24	351	-18	-20	0.33	12	24	-11	97	82
8	2011-02-15 04:08	10	-10	27	0.87	29	49	10	77	104
9	2011-09-06 23:39	29	20	-90	0.49	30	26	29	74	124
10	2012-01-23 04:39	19	41	64	0.77	55	37	9	99	123
11	2012-06-14 14:54	360	-28	11	0.90	30	53	1	116	117
12	2012-07-12 $17:54$	8	-12	68	0.46	30	26	14	106	129

obtained from the ecliptic cut is quite comparable to (within 5° of) the propagation direction from the GCS reconstruction. Only for CMEs #6 and #10 do we find a larger difference of close to 30° and 10°, respectively. The reason can be found in the combination of low/high latitude and large tilt angle. Therefore, the part within the ecliptic plane does not correspond well to the main propagation direction resulting from GCS reconstruction for these two CMEs.

3.2 WSA/HUX model

In the following paragraph, we summarize the main characteristics of the numer-230 ical framework used here for modelling the physical conditions in the evolving ambient 231 solar wind flow. For this study, we make use of the framework shown in Reiss et al. 232 (2019, 2020), but the components of this framework were developed by Wang and 233 Sheeley (1995), Arge et al. (2003), Riley and Lionello (2011), and Owens and Riley 234 (2017). Specifically, we use magnetic maps of the photospheric field from Global Os-235 cillation Network Group (GONG) provided by the National Solar Observatory (NSO) 236 as input to magnetic models of the solar corona. Using the Potential Field Source 237 Surface model (PFSS; Altschuler & Newkirk, 1969; Schatten, Wilcox, & Ness, 1969) 238 and the Schatten current sheet model (SCS; Schatten, 1971) we compute the global 239 coronal magnetic field topology. While the PFSS model attempts to find the potential 240 magnetic field solution in the corona with an outer boundary condition that the field 241 is radial at the source surface at 2.5 R_{\odot} , the SCS model in the region between 2.5 and 242 5 R_{\odot} accounts for the latitudinal invariance of the radial magnetic field as observed by 243 Ulysses (Wang & Sheeley, 1995). From the global magnetic field topology, we calcu-244 late the solar wind conditions near the Sun using the established Wang-Sheeley-Arge 245 (WSA) model. To map the solar wind solutions from near the Sun to Earth, we use the 246 Heliospheric Upwind eXtrapolation model (HUX) which simplifies the fluid momen-247 tum equation as much as possible. The HUX model solutions match the dynamical 248 evolution explored by global heliospheric MHD codes fairly well while having low pro-249 cessor requirements. An example of the ambient solar wind modeled by WSA/HUX 250 combination is shown in Figure 2. 251

3.3 ELEvoHI ensemble modeling

ELEvoHI uses HI time-elongation profiles of CME fronts and assumes an elliptical 253 shape for those fronts to derive their interplanetary kinematics (detailed information 254 about the underlying Ellipse Conversion method can be found in Rollett et al., 2016). 255 The tracking of each CME was done manually using ecliptic time-elongation maps (j-256 maps; Davies et al., 2009; Sheeley et al., 1999), generated by extracting ecliptic data 257 from STA and STB HI images. Transients, like CMEs, appear as a bright feature in 258 the j-maps. To extract the time-elongation profiles, we use the SATPLOT tool im-259 plemented in IDLTM SolarSoft. It allows any user to measure the elongation, which 260 is defined as the angle between the Sun - observer (STA or STB) line and the CME 261 front. ELEvoHI converts the resulting time-elongation profiles to time-distance pro-262 files, assuming an elliptic frontal shape using the ELEvoHI built-in procedure ELipse 263 Conversion (ELCon; Rollett et al., 2016). 264

ELEvoHI accounts for the effect of the drag force exerted by the ambient so-265 lar wind, which is incorporated in the model. The drag force is an essential factor 266 influencing the dynamic evolution of CMEs in the heliosphere. Within ELEvoHI, 267 the time-distance track is fitted using a drag-based equation based on the drag-based 268 model (DBM) given in Vršnak et al. (2013). The user has to define the start- and 269 end point for the DBM fit (usually around $30 - 100 \text{ R}_{\odot}$) in the time-distance profile. 270 In order to account for the de-/acceleration of the CME due to drag, an estimate of 271 the ambient solar wind speed is needed. Here we make use of the WSA/HUX model 272 (see Section 3.2). It provides the ambient solar wind conditions for a full Carrington 273 rotation (see Figure 2). We only consider the part of the full map according to the 274 start- and end-point selected by the user, and the CME propagation direction and half 275 angle from EAGEL. From this area, surrounded by the white box in Figure 2, we take 276 the median of the solar wind speed and define the uncertainties to be ± 100 km s⁻¹, 277 based on a study by Reiss et al. (2020). They considered nine years (mid 2006 to mid 278 2015) and report a mean absolute error of the WSA solar wind speed prediction with 279 respect to the in situ speed of 91 km s⁻¹. The obtained ambient solar wind speed with 280 its uncertainty is split into steps of 25 km s⁻¹, which gives nine different input speeds 281 to ELEvoHI. Based on these speeds, ELEvoHI selects the combination of drag param-282 eter and ambient solar wind that best fits the time-distance profile for each ensemble 283 member (for a detailed description see Rollett et al., 2016). 284

Since ELEvoHI is a 2D model, we are only interested in the propagation of a CME in the ecliptic plane. ϕ and λ , in this plane, are provided by EAGEL (see Section 3.1). The inverse ellipse aspect ratio, f, defines the shape of the assumed CME front in the ecliptic plane, where f = 1 represents a circular front while f < 1 corresponds to an elliptical CME front (with the semi-major axis perpendicular to the propagation direction)

To run ELEvoHI in ensemble mode, we vary ϕ , λ and f. A details description can 291 be found in Amerstorfer et al. (2018) and the code is available online (see Section 6). 292 ϕ and λ vary over a range of $\pm 10^{\circ}$ from their values obtained from EAGEL, with a 293 step size of 2° and 5°, respectively. This range is defined based on a study by Mierla et 294 al. (2010), who report an uncertainty in the parameters when different users manually 295 perform GCS reconstruction. For f we set a fixed range from 0.7 - 1.0 (0.1 step size). Thus we get a total of 220 ensemble members for one ELEvoHI event (i.e. 11 values 297 of ϕ , 5 values of λ and 4 values of f). For each ensemble member we select a different 298 sector from the ambient solar wind provided by the WSA/HUX model combination 299 300 according to the propagation direction, half angle, start- and end-point. In Figure 2, the WSA/HUX model results for event #5 are shown. The white box indicates the 301 area from which the ambient solar wind speed for one individual run of ELEvoHI is 302 computed. Shown is the area for the minimum propagation direction, ϕ_{STA} of 56° with 303 a λ of 50°. For each ensemble member the area surrounded by the white box is slightly 304



Figure 2: Ambient solar wind speed provided by the WSA/HUX model for event #5. The white box defines the area that is used to calculate an estimate of the ambient solar wind speed for the ensemble member of ELEvoHI corresponding to the minimum propagation direction ($\phi_{\text{STA}} = 56^{\circ}$) with the maximum half width ($\lambda = 50^{\circ}$). The black box indicates the total area based on all the ensemble members of ELEvoHI for this event. The longitude of 0° corresponds to the longitude of Earth.

different according to ϕ and λ . The black box plotted indicates the total area based on all ELEvoHI ensemble members for this event.

Running ELEvoHI in ensemble mode enables us to calculate a mean and a median predicted CME arrival time and also to define an uncertainty. In addition, we can give a probability for whether a CME is likely to hit Earth or not. When all of the 220 ensemble members predict an arrival at Earth, we assume the predicted likelihood of an Earth hit to be 100%.

312 4 Results

We perform ELEvoHI ensemble modeling for 12 CMEs between February 2010 to July 2012 (see Table 1) and compare the predicted arrival times based on STA and STB HI observations with each other. The CMEs propagated close to the ecliptic plane and showed clear in situ signatures at L1. A prerequisite for the chosen CMEs was that the CMEs could be tracked unambiguously in both STA and STB HI j-maps.

In Table 3, we list the predicted ensemble median arrival times and speeds with their standard deviation for each CME under study. It further contains the difference between the predictions from the two vantage points. We find that the predicted arrival times for STA and STB can deviate by up to 9.5 hrs while the mean difference is 6.5 hrs. The mean difference in the arrival speed is 63 km s⁻¹, with an exceptionally large discrepancy of 189 km s⁻¹ for event #10.

The largest arrival time differences are found for events #2 and #9. The arrival probability, based on the number of ensemble members that are predicted to hit Earth, is 79% for event #2 and only 56% for event #9. According to their relatively large angle of propagation with respect to the Sun-Earth line, the CMEs #2 ($\phi_{Earth} = 22^{\circ}$, $HA = 30^{\circ}$) and #9 ($\phi_{Earth} = 30^{\circ}$, $HA = 30^{\circ}$) are considered as "flank hits". In such cases, ELEvoHI tends to predict the CME arrival time to be later than expected.

Table 3: List of predicted median arrival times $(Date)$ and the standard deviation (SD)
based on STA and STB observations, respectively. STA - STB gives the difference be-
tween the predicted median arrival times. v is the predicted median arrival speed with the
standard deviation and $v_{\text{STA}-\text{STB}}$ is the difference in arrival speed between STA and STB
predictions

Nr.	Date STA	SD_{STA}	Date STB	SD_{STB}	STA - STB	$v_{\rm STA}$	$v_{\rm STB}$	$v_{\rm STA-STB}$
		[h]		[h]	[h]	$[{\rm km \ s^{-1}}]$	$[{\rm km \ s^{-1}}]$	$[{\rm km \ s^{-1}}]$
1	2010-02-07 11:24	1.5	2010-02-07 20:24	2.1	-9.0	455 ± 17	395 ± 11	60
2	2010-03-24 07:17	9.1	2010-03-24 16:40	4.1	-9.5	401 ± 32	351 ± 11	50
3	2010-04-05 13:23	2.5	$2\overline{0}1\overline{0}-\overline{0}4-\overline{0}5$ 16:06	0.4	-2.7	649 ± 37	625 ± 5	24
4	2010-04-11 16:07	0.6	2010-04-12 00:12	5.1	-8.1	443 ± 6	391 ± 33	52
5	$\overline{2010-05-27}$ 17:36	1.9	2010-05-28 02:26	1.2	-8.8	455 ± 9	407 ± 9	48
6	2010-10-30 11:24	1.4	2010-10-30 04:43	7.1	6.7	432 ± 7	476 ± 45	-44
7	2011-02-04 01:08	2.4	2011-02-03 22:24	7.3	4.5	387 ± 9	446 ± 34	-59
8	$\overline{2011}$ - $\overline{02}$ - $1\overline{8}$ 06:22	2.8	$2\overline{0}1\overline{1}-\overline{0}2-\overline{1}8$ 10:34	6.1	-4.3	478 ± 18	407 ± 50	71
9	$\overline{2011}$ - $\overline{09}$ - 10 18:55	14.9	2011-09-10 09:48	5.4	9.1	396 ± 46	430 ± 18	-34
10	2012-01-24 17:49	4.0	$2\overline{0}1\overline{2}$ - $\overline{0}1$ - $\overline{2}4$ 13:29	3.6	4.3	793 ± 103	982 ± 150	-189
11	$\overline{2012-06-16}$ 15:47	3.8	$2\overline{0}1\overline{2}$ - $\overline{0}6$ - $\overline{1}6$ 07:53	5.2	7.9	712 ± 72	749 ± 143	-37
12	$\overline{2012-07-14}$ 22:16	4.9	$2\overline{0}1\overline{2}$ - $\overline{0}7$ - $\overline{14}$ 18:53	3.7	3.5	658 ± 80	579 ± 28	89

The reason may be found in the assumed circular CME front for f = 1.0. For future versions of ELEvoHI, we will consider different approaches to tackle such extreme delays for flank encounters.

Event #11 occurred on June 14, 2012 and was studied e.g. by Kubicka et al. 333 (2016) who report two preceding CMEs. However, the WSA/HUX model does not 334 provide the ambient solar wind conditions with preceding CMEs included and is there-335 fore most probably not suitable for interaction events. The events #1, #4, and #5336 also show large differences in the predicted arrival times based on STA and STB obser-337 vations. However, these differences are most certainly related to large variance in the 338 modeled ambient solar wind speeds that are used as input to ELEvoHI (see Section 4.2 339 and Figure 5 and 6). 340

341

4.1 Tracking different parts of the CME front

It is important to keep in mind that different parts of the CME front are tracked 342 in STA and STB HI images. This leads to different input conditions to the ELEvo 343 propagation model for STA and STB. ELEvoHI is designed to take HI tracks for the 344 same CME from different viewpoints. Ideally, predictions should give the same CME 345 speed and direction in both cases. One problem is, however, that the CME is not 346 behaving as a single coherent entity, but is instead moving with different speeds at 347 different longitudes (Owens, Lockwood, & Barnard, 2017), which is not incorporated 348 within ELCon nor in any other HI conversion method (e.g. SSE, FP, and HM). 349

Figure 3 presents two snapshots of a movie for event #5, with the ambient solar wind provided by WSA/HUX model combination and the positions of various spacecraft and planets. The elliptical CME fronts from one ensemble member based on STA and STB observations are shown in red and blue, respectively. The green dots indicate the apex direction of those idealized CME fronts. Further, gray lines from





Figure 3: Two snapshots of the CME propagation for one ensemble member based on STA (red) and STB (blue) observations for event #7. The ambient solar wind is computed using the WSA/HUX model combination. The elliptical CME fronts from one ensemble member based on STA and STB observations are shown in red and blue, respectively. The green dots represent the position of the apex of the idealized CME fronts. The curved lines in red and blue show the intercept of the idealized elliptical front of the CME and the tangent (gray lines) for each time step over the course of the simulation for STA and STB, respectively. Link to the movie.



Figure 4: Ambient solar wind speed at the tangent points for event #7. Plotted are time series of the elongation angles of the tangent points as seen from STA (top panel) and STB (bottom panel) colour-coded according to the speed of the ambient solar wind at that tangent point.

the two STEREO spacecraft to the elliptical CME fronts are plotted. These tangents 355 correspond to the elongations of the leading edge of the CME at these times. At the 356 end of these lines, we add a point, which is the 'tangent point' at each time step. Over 357 the course of the simulation, these points trace out curved lines, in red and blue for 358 STA and STB, respectively. From Figure 3, it is obvious that, in the near-Sun part of 359 the HI FOV, the observed leading edge is close to the apex of the idealized CME front 360 for both STEREO spacecraft. As the CME propagates, the tangent point, i.e. the part 361 of the CME with the greatest elongation seen by STA and STB progressively moves 362 out to the flanks of the ellipse. Based on the observations of these tangent points, the 363 prediction for the whole front is conducted. Hence, the apex of the CME is, if at all, 364 only observed for a short period of time. In order to get an estimate of the CME Earth 365 arrival we have to assume a designated shape of the CME front, which is in our case, 366 an ellipse. As shown by Owens et al. (2017) this assumption might not be valid since 367 the CME interacts with the ambient solar wind. 368

369

4.2 Effect of the ambient solar wind

When considering different points along the idealized elliptical CME front, it is noteworthy that the ambient solar wind speeds at these points would likely be different. Furthermore, the part of the CME front corresponding to the greatest elongation as seen by STA and STB (i.e. the points corresponding to the tangent to the CME front) would propagate in different ambient solar wind conditions. In Figure 4, the modeled time-elongation profiles of the tangent points seen from STA (top panel) and STB (bottom panel) are shown.

These profiles are obtained from one modeled ensemble member of the ELEvoHI prediction, separately for STA and STB (see Figure 3), and are therefore available from 2011-01-30 23:00 until 2011-02-06 11:00. As long as the CME front could be tracked in HI images (until about 20110202 01:00), the plotted profiles are consistent with the measured HI time-elongation profiles, obtained using the SATPLOT tool. The colors represent the speed of the ambient solar wind at the corresponding points. ³⁸³ Due to the propagation of the modeled CME in the heliosphere, the elongation of the ³⁸⁴ tangent point ranges from roughly 6° to about 64° and the speed of the ambient solar ³⁸⁵ wind at these points ranges from 470 km s⁻¹ to 525 km s⁻¹ for STA (top panel in ³⁸⁶ Figure 4). The range of the elongation is slightly larger for STB (5° to 74°) but the ³⁸⁷ ambient solar wind speed ranges only from \approx 380 km s⁻¹ to \approx 480 km s⁻¹.



Figure 5: Ambient solar wind speed provided by the WSA/HUX model combination for all 12 events under study. The black boxes define the areas that are used to estimate how structured the ambient solar wind is for each CME. Longitude of 0° corresponds to the longitude of Earth.

In the previous paragraph, we considered the ambient solar wind speed at the 388 tangent point for one ensemble member. Additionally, we examine the distribution of 389 the ambient solar wind speed considered for all ensemble members (see black boxes in 390 Figures 2 and 5) that are used as input to ELEvoHI for a single CME. From the areas 391 framed by the black boxes, we calculate the standard deviation and correlate those to 392 the absolute values of the difference between STA and STB arrival time predictions 393 for each event (see Figure 6). This gives us the possibility to check the influence of the 394 ambient solar wind on the arrival time differences. We obtain a Pearson correlation 395 coefficient of cc = 0.52 for all events under study. However, when excluding events #2 396 and #9, which are considered as "flank hits", and excluding event #11 (CME-CME 397 interaction event), the Pearson correlation coefficient increases to cc = 0.92. This 398



Figure 6: Standard deviation of the ambient solar wind vs. the arrival time difference between STA and STB predictions. The Pearson correlation coefficient for all events under study (black) is calculated. In blue we present the Pearson correlation coefficient and a linear fit when excluding the outliers (indicated by the red boxes), i.e. flank hits (Events #2 and #9) and the CME-CME interaction event (Event #11).

indicates that a more structured ambient solar wind (i.e. a larger standard deviation)
 leads to a larger differences between STA and STB arrival time prediction.

401 4.3 Comparison to in situ arrivals

Figure 7 shows the distributions of the arrival time and arrival speed differences 402 with respect to the in situ arrivals for all ensemble members for each CME. Blue and 403 orange correspond to STB and STA ensemble predictions, respectively. The black 404 horizontal lines indicate the median values of each distribution. When comparing 405 the median predicted arrival times to the in situ arrivals, we obtain a mean absolute 406 error (MAE) over all events of 7.5 \pm 9.5 hrs and a root mean square error (RMSE) of 407 ≈ 10.4 hrs. A mean error (ME) of ≈ 4 hrs indicates, in this setup, that ELEvoHI tends 408 to predict the arrivals too late. The highest arrival time discrepancy is found for event 409 #9 where the prediction based on STA is 31 hrs too late. When comparing the median 410 predicted arrival speeds to the in situ speeds we get a MAE of 87 ± 111 km s⁻¹, a 411 RMSE of ≈ 123 km s⁻¹ and a ME of ≈ 52 km s⁻¹. The highest speed difference is found 412 for the STB prediction of event #10, overestimating the arrival speed by 369 km s⁻¹. 413

Interestingly, event #10 gives an accurate predicted arrival time, even though the 414 predicted arrival speed is highly overestimated. When performing GCS reconstruction 415 we obtain a high latitude and a large tilt angle for this CME meaning that the 3D 416 propagation direction differs from that in the ecliptic plane (see Table 2). As already 417 mentioned, event #11 is a CME-CME interaction event which explains the large dis-418 crepancy especially for the predicted arrival speed. The reason might be found in an 419 extremely low drag due to preconditioning in the interplanetary space (Y. D. Liu et 420 al., 2014; Rollett et al., 2014; Temmer & Nitta, 2015). 421



Figure 7: Frequency distributions derived from all ensemble members for the arrival time prediction (top panel) and the arrival speed prediction (bottom panel) based on HI data from STB (blue) and STA (orange), respectively. In the top panel, positive values correspond to a late arrival time prediction while negative values indicate an early arrival prediction. Positive/negative values in the bottom panel indicate an over-/underestimated arrival speed prediction. The black horizontal bars show the median values of the distributions of all the ensemble members for STB and STA.

5 Discussion and Conclusions

We present the ELEvoHI ensemble modeling results for 12 CMEs, occurring be-423 tween February 2010 and July 2012, that were observed by both STEREO spacecraft. 424 This study mainly focuses on the difference of the modeled arrival time and arrival 425 speed when using STA and STB HI observations, separately. We find on average a 426 difference of 6.5 hrs between arrival time predictions from the two spacecraft but the 427 largest difference is about 9.5 hrs for event #9. For the arrival speed we find a mean 428 difference between STA and STB predictions of 63 km s^{-1} with a maximum difference 429 of 189 km s⁻¹ for event #10. 430

ELEvoHI tends to predict the arrival time later than observed for CMEs that are considered as 'flank hits' (event #2 and event #9). For such events the propagation direction with respect to Earth is larger than 20°, and not all of the ensemble members predict an Earth impact. The reason for the late arrival prediction may be found in the assumed circular shape (for f = 1.0) and the highly curved flanks.

We provide two CME arrival time and arrival speed predictions, from STA and 436 STB observation, for the same CME to examine the reasons for the discrepancy be-437 tween these two predictions. We find, that the CME front propagates in different 438 ambient solar wind conditions when observed in STA and STB HI images. However 439 the kinematics of the CME front obtained e.g. by STA data is used for modeling of the 440 whole CME front, including the Earth-directed part. The same applies for predictions 441 based on STB data, which is the reason for the differences in the predicted arrival 442 times based on STA and STB observations. 443

We further see, that an ambient solar wind exhibiting a high variance within the area used for ELEvoHI model predictions leads to larger discrepancies between STA and STB model predictions. We obtain a Pearson correlation coefficient (cc = 0.92), when excluding flank hits (events #2 and event #9) and the CME-CME interaction event (event #11). Furthermore, we assume that in such cases the CME front is more likely to deform from an idealized elliptical shape due to interaction with the ambient solar wind (Owens et al., 2017; Riley & Crooker, 2004).

The current CME forecasting abilities in the community are summarized in Riley 451 et al. (2018). They analyzed CME forecasts that have been submitted to the Commu-452 nity Coordinated Modeling Center (CCMC) scoreboard from 2013 to mid-2018. The 453 CCMC scoreboard is a platform provided to scientists to compare their forecasts with 454 each other in real-time. Riley et al. (2018) found that the CME shock arrival times 455 for all models combined are predicted on average within ± 10 hrs but with standard 456 deviations of sometimes more than 20 hrs. The best model performance was found 457 for the WSA-ENLIL+Cone model (Odstrcil et al., 2004), run by the UK Met Office, 458 having a bias of 1 hour, a MAE of 13 hrs and a standard deviation of 15 hrs. The 459 results of this study are similar to the findings of Riley et al. (2018) when comparing 460 the modeled arrival times to the actual arrivals of CMEs, as determined from in situ 461 measurements. Here, we only perform hindcasts of CME arrivals. For the 24 arrival 462 predictions (12 based on STA and 12 based on STB observations), we obtain a MAE 463 of 7.5 \pm 9.5 hrs, a RMSE of \approx 10.4 hrs and a ME of \approx 4 hrs for the arrival time. For 464 the arrival speed, we get a MAE of 87 ± 111 km s⁻¹, a RMSE of ≈ 123 km s⁻¹ and a 465 ME of ≈ 52 km s⁻¹. 466

⁴⁶⁷ As already mentioned, event #11 is a CME-CME interaction event studied e.g. by ⁴⁶⁸ Kubicka et al. (2016). This CME was closely preceded by two other CMEs that erupted ⁴⁶⁹ one and two days before this event and that altered the conditions in the heliosphere. ⁴⁷⁰ The arrival time prediction for this CME is about 11 hrs too early, while the arrival ⁴⁷¹ speed is greatly overestimated (by 260 km s⁻¹) using the ambient solar wind solutions ⁴⁷² provided by the WSA/HUX model. However, this model does not consider preceding

CMEs and is likely not valid in such cases. An additional approach to infer the ambient 473 solar wind conditions in the low heliosphere is shown in Barnard, Owens, Scott, and 474 Jones (2019). In this study the authors established a statistical relationship between 475 the solar wind speed in the low heliosphere and the variability in HI images. A recent 476 study by Amerstorfer et al. (2020) focuses on different input parameters to ELEvoHI 477 including three possible methods to infer the ambient solar wind conditions needed 478 by the model. Their results indicate that the ambient solar wind obtained from the 479 WSA/HUX model provides the best results. 480

481 ELEvoHI provides ensemble predictions based on various inputs, namely propagation direction, half width, inverse aspect ratio and ambient solar wind speed. In the 482 current version, ELEvoHI is not able to react to possible deflections of a CME during 483 its propagation. Furthermore, the elliptical CME shape, once defined by the input 484 parameters, does not change during propagation. This has been shown to be invalid 485 by, for example, Rollett et al. (2014), who performed a case study by combining HI 486 data with in situ data to ascertain the kinematics of the 2012, March 7 CME. The 487 authors demonstrated evidence for an asymmetric evolution of the CME, which was 488 caused by the preconditioned ambient solar wind resulting in a different drag regime 489 influencing different parts of the CME. 490

Barnard et al. (2017) found that the failure to take into account CME deflection and deformation is an important factor when considering CME propagation in the heliosphere and would likely lead to uncertainties in the arrival time and arrival speed prediction. The authors additionally showed that different tracks lead to quite different CME arrival time predictions. By using HI observations with better solar wind modeling and varying CME frontal shapes we should be able to improve our current arrival time predictions (Barnard, Owens, Scott, & de Koning, 2020).

A number of studies have taken advantage of stereoscopic HI observations, from 498 the two STEREO spacecraft, to glean information on CME propagation and evolution 499 (e.g. Davies et al., 2013; Y. Liu et al., 2010; Lugaz, 2010; Volpes & Bothmer, 2015). 500 We believe, that a stereoscopic view on CMEs incorporated in ELEvoHI will improve 501 the arrival time predictions substantially. Therefore, we strongly support ESA's L5 502 mission, equipped with a heliospheric imager (Kraft, Puschmann, & Luntama, 2017; 503 Lavraud et al., 2016), and an additional heliospheric imager at L1. Fortunately, the 504 upcoming Earth-orbiting PUNCH mission (launch planned in 2023) will also possess 505 wide-angle white-light heliospheric imagers, as well as a coronagraph, and will be 506 able to provide additional observations of CMEs. Based on information from these 507 additional vantage points, more accurate CME arrival predictions are likely to be 508 achieved. Since ELEvoHI is ready to be used in near real-time, future HI observations 509 are essential for further CME arrival predictions. STA, currently near L5, will have 510 moved beyond L4 by 2027, so it will be necessary to have heliospheric imagers that 511 are observing the space between Sun and Earth after around 2030. 512

In a next step, we want to further develop ELEvoHI in such a way that it can 513 combine HI data from two vantage points in order to constrain the CME and exclude 514 ensemble runs that are not consistent with the observations. Also the CME shape 515 can be constrained by multiple HI observations and therefore, we aim to make the 516 CME front deformable during the propagation through the heliosphere. Hence, the 517 assumed elliptical CME front would be able to adjust according to the ambient solar 518 wind conditions. Scott et al. (2019) showed that ghost fronts in the HI observations 519 can be used to infer the structure of a CME. Using their approach, we also aim to 520 521 improve our model by verifying and constraining the CME shape.

522 6 Data Sources

523 Data

- 524 STEREO/HI: https://www.ukssdc.ac.uk/solar/stereo/data.html
- 525 STEREO/COR2 and SoHO/LASCO: https://sdac.virtualsolar.org/cgi/
- 526 NSO/GONG: https://gong.nso.edu/data/magmap/
- 527 HELCATS: https://www.helcats-fp7.eu
- ⁵²⁸ ICMECAT: https://doi.org/10.6084/m9.figshare.6356420

529 Model

- 530 ELEvoHI is available at https://zenodo.org/record/3873420.
- EAGEL is available at https://zenodo.org/record/4154458.

532 **Results**

- ⁵³³ The visualization of each prediction result, i.e. movies and figures, can be downloaded
- ⁵³⁴ from https://doi.org/10.6084/m9.figshare.12758312.v1.

535 Software

- ⁵³⁶ IDLTM Version 8.4
- 537 Python 3.7.6
- 538 SATPLOT: https://hesperia.gsfc.nasa.gov/ssw/stereo/secchi/idl/jpl/satplot/
- 539 SATPLOT_User_Guide.pdf

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