RENU2 Rocket Observations of Fine-Scale Thermal Ion Upflow, Downflow, and Temperature

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Abstract

We present an analysis of in-situ thermal ion measurements from a cusp auroral sounding rocket. Using a forward modeling procedure, we find most-probable thermal ion temperature and parallel (field-aligned) bulk flow velocity along the trajectory. Spatially and temporally intermittent fine-scale structure in upflowing/downflowing features in the dayside cusp ionosphere are presented. We show that the observed ion temperatures are consistent with Joule heating expectations if spatially and temporally intermittent drivers and responses in the dynamic cusp environment are considered. Additionally, a forward modeling procedure for the ion data interpretation is improved and a sensitivity analysis is presented.

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21	Key points
22	• Temporally intermittent drivers are required to model the observed thermal ion
23	temperatures consistently with Joule heating.
24	• Spatially intermittent signatures are observed of ion upflow amidst overall down-
	four besting and flow regions are stripted and adjacent
25	now, nearing and now regions are structed and adjacent.
26	• A Maxwellian forward modeling procedure for the ion data interpretation is im-

proved and a sensitivity analysis is presented.

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28 Abstract

We present an analysis of in-situ thermal ion measurements from a cusp auroral 29 sounding rocket. Using a forward modeling procedure, we find most-probable thermal 30 ion temperature and parallel (field-aligned) bulk flow velocity along the trajectory. Spa-31 tially and temporally intermittent fine-scale structure in upflowing/downflowing features 32 in the dayside cusp ionosphere are presented. We show that the observed ion temper-33 atures are consistent with Joule heating expectations if spatially and temporally inter-34 mittent drivers and responses in the dynamic cusp environment are considered. Addi-35 36 tionally, a forward modeling procedure for the ion data interpretation is improved and a sensitivity analysis is presented. 37

Keywords (auroral ionosphere, cusp ion upflow, dayside cusp thermal plasma, spa tial and temporal intermittency)

40 **1** Introduction

At the low altitudes of ionospheric sounding rockets ($\approx 200 \,\mathrm{km}$ to $450 \,\mathrm{km}$), the pre-41 dominant field-aligned motion of the ambient thermal ion population in auroral regions 42 can be downgoing, and may indicate signatures of flux tubes with upflow at higher al-43 titudes, or with upflow at previous times (e.g. Burleigh et al., 2019). This has been seen 44 both as a downflowing vertex of energetic ion conics (Arnoldy et al., 1996), and as an 45 overall downflow of the bulk thermal ion population as discussed here for the dayside case, 46 on the nightside by Fernandes et al. (2016), and commonly throughout the polar cap by 47 DE-2 (Loranc et al., 1991). 48

The Rocket Experiment for Neutral Upwelling (RENU2) was launched on Decem-49 ber 13, 2015 at 07:34UT from the Andøya Space Center (Norway) into a neutral upwelling 50 event to study particle behavior between 200 km and 450 km (Lessard et al., 2019). Us-51 ing a forward modeling procedure (Fernandes et al., 2016; Fernandes & Lynch, 2016) and 52 calculating Maxwellian distributions constrained by other diagnostics (e.g. GPS veloc-53 ity, in situ DC electric field, radar electron density profiles, and in situ ambient electron 54 temperature (T_e) for the spacecraft's sheath potential, denoted ϕ_{sheath}), we find the Maxwellian 55 parameters that best compare to distribution function 2-D slices observed by a thermal 56 ion electrostatic analyser, establishing a database of most-probable thermal ion temper-57 ature (T_i) and parallel (field-aligned) bulk flow velocity $(v_{i_{\parallel}})$ along RENU2's trajectory. 58 In the Discussion section, we interpret these observations in the context of the Geospace 59 Environment Model of Ion-Neutral Interactions with Transverse Ion Acceleration (GEMINI-60 TIA) 2-D ionospheric model (Burleigh & Zettergren, 2017), showing that intermittency 61 in both drivers and responses is necessary for interpreting this event. 62

63 2 Instrumentation

We present data from the HEmispherical Energy Particle Spectrometer (HEEPS-64 T) (Fernandes et al., 2016), an electrostatic analyser which measures the ionospheric ther-65 mal core (energies 0.12 - 22 eV in the spacecraft frame). We use the HEEPS-T 2-D slices 66 of the ion distribution functions to calculate temperature and parallel bulk flow veloc-67 ity. While HEEPS-T data are collected at 8 Hz per energy-pitch angle frame, the ther-68 mal ion population is heavily non-gyrotropic in the payload frame. We concentrate here 69 on the four or five slices (at an $8 \, \text{Hz}$ cadence) each payload spin period $(1.8 \, \text{s})$ where the 70 optimal side of the HEEPS-T aperture is ram-looking. Our calculation accounts only for 71 O^+ (16 amu). 72

The on-board GPS provides RENU2's position and velocity. The electric field instrument was mounted on a nearby ($\approx 150 \,\mathrm{m}$ from the main payload) sub-payload dur-

ing RENU2's flight. Using 12 m tip-to-tip wire booms, and sampling at 1kHz, the elec-75 tric field instrument provides high-fidelity in-situ measurements of the DC and wave elec-76 tric fields (Klatt et al., 2005). These measurements are used to derive the F-region plasma 77 $\mathbf{E} \times \mathbf{B}$ drift velocities, presented here in the Earth-Centered-Earth-Fixed (ECEF) frame, 78 using the attitude solution and the payload's (GPS) ram velocity to translate and ro-79 tate observations from the instrument frame into the ECEF frame. The Ionization Gauge 80 (IG) instrument (Clemmons et al., 2008) consists of a set of pressure gauges that mea-81 sure thermospheric gas pressure characterizing in-situ neutral winds. As with the plasma 82 bulk flow, these neutral winds are presented here in the ECEF frame with payload mo-83 tion removed. 84

The Electron Retarding Potential Analyzer (ERPA) provides an estimate of ϕ_{sheath} by measuring the ambient thermal electron temperature. ϕ_{sheath} is interpreted here as $5k_bT_e$, where k_b is the Boltzmann constant, consistent with current balance expectations when sunlight and work function effects cancel each other out. (Frederick-Frost et al., 2007; Siddiqui et al., 2011). The Electron PLASma instrument (EPLAS) provides measurements of electron precipitation (Kenward et al., 2020).

Electron data from the European Incoherent SCATter (EISCAT) Svalbard 42 m 91 radar (ESR) indicated multiple transients in temperature prior to launch, consistent with 92 poleward moving auroral form (PMAF) activity. RENU2 was launched into the fourth 93 of these PMAFs. Weak ion upflow can be observed in the ESR field-aligned profiles (data 94 available at spdf.gsfc.nasa.gov) above 400 km throughout the interval. Between 07:34 95 and 07:48 UT, upflow is observed in the ESR signatures down to 300 km altitude; near 96 07:50 UT, downflow is seen reaching down to 250 km altitude. RENU2 entered a region 97 of auroral precipitation at approximately T+450 s, near the 448 km apogee. 98

RENU2's ground track and the ESR field-aligned look direction are not common 99 volume (though both show the same cusp event). Hence, to support the thermal ion anal-100 ysis below, an altitude and geomagnetic activity dependent proxy for plasma density is 101 generated from radar data as a function of altitude and of auroral precipitation activ-102 ity level (minimal, low, and high as seen by EPLAS). When the EISCAT Very High Fre-103 quency (VHF) radar (based in Tromsø) is colocated with RENU2's trajectory (eg when-104 ever the payload is within 200 km (or, $\approx 2^{\circ}$ latitude; the typical width of the cusp au-105 rora, as on the upleg), these EISCAT VHF (Tromsø) data are used instead of ESR. Note 106 that RENU2's closest approach to the EISCAT VHF (Tromsø) field of view occurred at 107 T+378 s. The on-board ion observations provide a fine-scale view of the ion profiles near 108 the event seen at larger scales by ESR (16° E longitude), and somewhat to the east (RENU2's 109 flight trajectory spanned 16° E to 32° E longitude) within the same cusp event (trajec-110 tory detailed in Lessard et al. (2019)). 111

112 3 Maxwellian Distribution Calculations and Sensitivity Analysis

HEEPS-T measures 2-D slices of 3-D plasma distributions. Parameter-matching, 113 using a forward-modeling procedure Fernandes et al. (2016), is utilized to characterize 114 in-situ ion temperatures and velocities that produce the distribution. Here, we have made 115 the assumption that the distribution is Maxwellian, which is valid at rocket altitudes (\approx 116 $200 \,\mathrm{km}$ to $450 \,\mathrm{km}$), in the collisional ionosphere, for the thermal core of a distribution 117 (Fernandes et al., 2016)). For this analysis, we assume an isotropic distribution (eg. anisotropy 118 = 1). A previous modeling study of the RENU2 campaign predicts an anisotropy range 119 from 0.9 to 1.1 (Burleigh et al., 2019). 120

First, as outlined in Fernandes et al. (2016), known parameters are constrained at each time step, including the plasma motion relative to RENU2 (calculated from DC electric field (DCE) data, GPS data, and the attitude solution), ϕ_{sheath} (from ERPA), and plasma density (from the radar data proxy). Then, an empirically determined range of possible values for the ion temperature and parallel bulk flow velocity are allowed. Next,
3-D Maxwellian distributions are generated using the known parameters together with
permutations of the possible ion temperature and parallel bulk flow velocity. These 3D Maxwellians are translated and rotated into the payload frame, and then sliced in the
2-D plane consistent with the known attitude of the payload-mounted detector aperture
at each given time step. This slice is limited to the energy range of interest (above payload potential and saturation limits, and below the top energy step of HEEPS-T).

For each possible Maxwellian at each time step, the difference (of $\log_{10}(J_E)$, where 132 133 J_E is the differential energy flux; see Figure 2), between the forward-modeled Maxwellian and the data at each energy-pitch angle bin is calculated. The net summed squared dif-134 ference for each possible Maxwellian slice is accumulated as a metric, and the optimal 135 ion temperature and bulk flow velocity are selected by minimizing this metric for each 136 time step. We note that this mean-squared difference metric provides an improved pa-137 rameter search with less-scattered results than does the moment-based metric used in 138 Fernandes et al. (2016). 139

Figure 1 (a) - (d) shows examples of this comparison between calculated Maxwellians and observations, for a time equatorward of the auroral precipitation (T+400.6 s, panels a and c) and for another time within the precipitation region (T+458.7 s, panels b and d). These energy-pitch angle slices are shown in the sensor frame. The output of this parameter matching at each time step (with the GPS-measured payload velocity component along the field line removed) provides the temporal profiles of $v_{i_{\parallel}}$ and T_i discussed in the next Section.

The sensitivity of this forward-modelling calculation to its input plasma density 147 is illustrated by Figure 1 (e), (f). Plasma density is the most poorly-specified input pa-148 rameter, hence, we focus on sensitivity to that parameter. The ESR data along the rocket 149 trajectory upleg (not shown here) generally have higher densities than EISCAT VHF (Tromsø) 150 data, since ESR was under the cusp aurora soft precipitation activity, and the EISCAT 151 VHF (Tromsø) field of view was located in the subauroral zone, generally looking at the 152 low densities of the plasma trough. Using data proxies from ESR for the total flight pro-153 file (compared with the more closely colocated EISCAT VHF (Tromsø) data for times 154 before T+540 s) leads to an overestimate in the input plasma density by as much as a 155 factor of 2. If used as a forward-modeling input parameter, this high density results in 156 a reduction in the extracted ion temperature by 25% to 30% compared with the EIS-157 CAT VHF (Tromsø) driven data run, as seen in the T+300 s to T+450 s region of Fig-158 ure 1 (f). The red and blue curves illustrate the effect on the extracted T_i of deliberately 159 changing the best estimate of the density by 20%. An increase(/decrease) in the elec-160 tron density input into the forward model results in a decrease (/increase) in the extracted 161 ion temperatures. 162

¹⁶³ 4 Observations

Figure 2 shows the payload altitude and plasma environment (a) - (c), the forward-164 model-extracted thermal ion parallel-to- $\mathbf{B} v_{i_{\parallel}}$ (with the payload GPS velocity removed) (d), and the forward-model-extracted T_i (e), all vs. flight time. The optimally matched 165 166 $v_{i_{\parallel}}$ and T_i parameters shown here (and in Figure 1 (f) and in Figure 3 (b), (c)) have been 167 smoothed using a seven-point moving average, roughly a 2-spin-period (3.6 s) window. 168 Given a roughly 2 km/s payload velocity, this smoothing corresponds to 7 km resolution 169 for static spatial structure. The parallel bulk flow velocity quantifies the ion parallel-to-170 **B** upflow (negative) and downflow (positive, along **B**) present along the local magnetic 171 field. Note the prevalence of downflow with intermittent regions of localized upflow. Equa-172 torward of the precipitation (times before $T+425 \,\mathrm{s}$), the ions are mostly cold ($\approx 0.06 \,\mathrm{eV}$) 173 and downflowing ($\approx 400 \,\mathrm{m/s}$). 174



Figure 1: Maxwellian calculations: Panels (a) - (d): Parameter fitting: Measurements from HEEPS-T (a), (b) vs. the corresponding best-parameterized Maxwellian calculations (c), (d), with energy-pitch angle images presented in the sensor frame. Panels (e), (f): Forward-modelling sensitivity study: (e) different density estimates used as known inputs for the Maxwellian forward model calculation; (f) sensitivity of the algorithm to the input plasma density.



Figure 2: Stack-plot of thermal ion data in context of other diagnostics. Panel (a): in situ-observed DCE, and E' (described in text) (**E** in the ECEF frame, and E' in the neutral frame); (b) thermal ion J_E from HT for all pitch angles, in the sensor frame; (c) precipitating electron J_E from EPLAS; (d) extracted parallel-to-**B** bulk ion flow from HT, where positive values represent downflow (with the component of payload velocity along the field line removed); (e) extracted thermal ion temperature from HT, and T_i from EISCAT VHF (Tromsø).

Inside the auroral precipitation region (after T+450 s) the temperature data indicate the presence of colder temperatures interleaved with hotter, and the parallel flow velocity data indicate the presence of downflow interleaved with brief intervals of upflow. Not all hotter regions are upflowing, however. There is no local relationship between the upflow/downflow and the concurrent PMAF electron flux data (or the ELF spectral data, not shown here), as the timescales of electron precipitation and ion transport are very different.

182 5 Discussion

We comment on a few aspects of these observations. (1) The observed ion temperatures are only consistent with expectations for Joule heating processes if temporal and spatial variability are taken into account. (2) The prevalent parallel bulk flow velocity is downward, with brief, intermittent regions of upflow; and the regions of upflow and downflow are striated and nearby each other. (3) Rigorous modelling of such a scenario requires that temporal and spatial intermittency of both drivers and responses be considered. First, (1), let us consider the extracted ion temperatures in the context of the environment. A common expectation for ionospheric ion temperatures is that they are related to collisional heating between the neutral and ion populations. This is typically quantified by the following relation (Schunk et al., 1975; Fernandes et al., 2016):

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$$\mathbf{E}' = (\mathbf{E} + \mathbf{u}_{\mathbf{n}} \times \mathbf{B}) \tag{1}$$

$$T_i = T_n + 33.0 \cdot {E'}^2 \tag{2}$$

Here T_i and T_n (in eV) are the ion and neutral temperatures respectively, E (in V/m) 196 and **B** (in T) are the electric and magnetic fields, and $\mathbf{u_n}$ (in m/s) are the neutral ve-197 locities. We use here a magnetic field magnitude of 45,000 nT downward in the ECEF 198 frame for the full flight duration. The neutral wind velocities are 608 m/s east, 690 m/s199 north throughout the flight. These values provide an upper limit for the neutral winds 200 along RENU2's trajectory. These are much larger than the observed in-situ relative $\mathbf{E} \times$ 201 \mathbf{B} plasma drift velocities (in the ECEF frame), which rarely exceed 100 m/s during this 202 event. The magnitude of \mathbf{E}' (equation (1)) is shown (in gold), compared with the com-203 ponents of E (orange and blue), in Figure 2(a). The observed T_n (from the IG) ranges 204 from approximately 0.04 eV to 0.08 eV along RENU2's flight, with the average neutral 205 temperature being approximately 0.065 eV. 206

Figure 3(a) shows that extracted ion temperatures are roughly bounded between 207 $T_n + 33.0 \cdot E'^2$ and T_n . Allowing for time variable histories of **E** and **E'**, this result is 208 consistent with expectations from Joule heating processes. Note that only the average 209 observed value for each neutral parameter is used throughout the flight and there exists 210 no local point-to-point correlation between T_i and E'^2 as was seen previously on the night-211 side, where the DCE was the dominant factor (Fernandes et al., 2016). While neutral 212 wind parameters vary slowly, plasma DCE can vary abruptly and over sharp boundaries. 213 The in situ DCE, while of high fidelity and fine resolution, only indicate the state of the 214 DCE at the time of the rocket passage; the radar data indicate that higher DCE values 215 prevailed just before the rocket flight. If the earlier DCE and resulting $\mathbf{E} \times \mathbf{B}$ veloci-216 ties were comparable and parallel to the neutral wind flow such that E'^2 would have been 217 negligible (i.e., ions at rest in the neutral frame), the relative velocities in the neutral 218 frame would be small and frictional heating would be minimized, making T_n a reason-219 able lower bound for T_i (given the long time history of ion heating and upflow processes). 220 The average T_n throughout the flight was $0.065 \,\mathrm{eV}$ as shown in the figure, but it varied 221 between 0.04 eV and 0.08 eV. Deep within the auroral region where the plasma flow is 222 seen to be small, the relative velocity in the neutral frame is large, and Equation 2 pro-223 vides an upper bound for expected T_i . As the accompanying radars indicate a time-variable 224 plasma flow, the range of frictional heating expected varies from very little (at times when 225 the plasma flow may have been comparable to the neutral flow, i.e., T+350 s to T+425 s), 226 to very large (at the time of the rocket passage, i.e., T+500 s, where the weak plasma 227 flows are very different from the strong neutral winds), such that the observed thermal 228 ion temperatures can be consistent with Joule heating expectations. 229

Next, (2), we consider the prevailing downward bulk velocity of the thermal ions, 230 that contains localized and intermittent regions of upflow. Equatorward of the auroral 231 electron precipitation, the thermal ions are all downflowing. It is worth noting that suf-232 ficiently strong northward neutral winds, such as the $>500 \,\mathrm{m/s}$ northward component 233 of neutral wind observed here, can also act to suppress ion upflow, even to the point of 234 causing downflow (Burleigh & Zettergren, 2017). In the case of the RENU2 observations, 235 a 690 m/s northward neutral flow could account for 80-100 m/s of the bias toward down-236 ward field-aligned flow shown in Figure 2(d), given the difference between the field line 237 and local vertical along the trajectory ranging from 10 deg to 6 deg. 238



Figure 3: Panel (a): Extracted ion temperature compared with frictional heating expectations derived from in-situ DCE, and from neutral wind observations, see text. Panels (b), (c): Close-up of fine-structure in the extracted ion parameters, as discussed in Point (2). Data shown in this closeup panel include unsmoothed data points (orange) averaged over 4 samples, roughly a one-spin (1.8 s) cadence.

Within the cusp precipitation, there are localized embedded regions of upflow within 239 this downflow. Figure 3 (b), (c) details the localized transitions from a cold, downflow-240 ing region (T+535.6 s to T+546.8 s) to a warmer, upflowing region (T+546.8 s to T+550.6 s), 241 followed by another downflowing region (T+550.6 s to T+554.3 s). In this closeup plot 242 we show both the seven-point-moving-average values (blue) together with four-point-average 243 values with a one-spin (1.8 s, 4 km) resolution. Assuming Doppler-shifted spatial struc-244 turing, the first region is approximately 26 km in extent, the second region is approxi-245 mately 9 km in extent, while the third region is approximately 7 km in extent. Note the 246

narrow confinement of the heated upflowing ion structures, and their adjacence to colder
downflowing structures. These sharp transitions motivate the spatially bounded localized driver explored in the point (3) below, given the time needed to change ion population parameters.

Downflow at these altitudes can be an indication of upflow either at higher alti-251 tudes, or at earlier times. As an aside, we note that another thermal ion instrument on 252 RENU2, which roughly separates H^+ from heavier ions, shows a clear localized proton 253 downflow event between T+346 and T+348 s. This can be interpreted as the result of 254 255 a previously heated flux tube that was previously within the cusp activity, which at the time of the rocket passage is found just equatorward of the existing precipitation bound-256 ary. If the flux tube was recently heated by activity, but at the time of the observation 257 the driver has moved away, as is often the case with PMAFs, the H^+ would return to 258 lower altitudes after having failed to receive the energy necessary to cause it to outflow. 259 This proton-specific, localized event is a case for future study. 260

Thirdly, (3), we discuss the spatial and temporal intermittency required, for both 261 drivers and responses, to quantify an event such as seen by RENU2. Loranc et al. (1991), 262 in interpreting DE-2 observations of vertical flows at high latitudes, put their observa-263 tions in the context of localized, convecting flux tubes of frictional heating. Here we in-264 vestigate quantifying such a scenario specifically for the localized regions of upflow seen 265 in the RENU2 observations, using the GEMINI-TIA model (Burleigh & Zettergren, 2017). 266 The time evolution required for ion heating and upflow, coupled with the small scales 267 of the localized regions, means that the drivers and the responses may not be co-located 268 in time or space. However, the sharp boundaries observed make inferences about spa-269 tial structuring reasonable, and the large temperature changes observed constrain mod-270 elled parameters. Figure 4 illustrates one such scenario, where a localized flux tube (0.3 deg)271 wide) is driven with frictional heating imposed by a localized region of DCE $(40 \,\mathrm{mV/m})$ 272 for a short period of time (60 sec). The driver is sufficient to raise the local ion temper-273 ature to the observed values, and the hysteresis on the flux tube allows the effects of this 274 275 heating to remain for 30-60 s after the driver is turned off.



Figure 4: A GEMINI-TIA simulation of spatially and temporally intermittent frictional heating. A DCE of 40 mV/m is applied over a 0.3 deg region for 60 seconds (from T+1 min to T+2 min.) The upper left panel shows the heated flux tube at T+1.5 min, and the lower left panel shows profile cuts at 350 km altitude at a sequence of times. The heating region remains locally confined. The right panels show the time evolution along the center of the heated flux tube. The heating moves up from the E-region to fill the entire flux tube within a few 10s of sec, and remains at the upper end of the flux tube for up to a minute after the driver is removed.

²⁷⁶ 6 Conclusions

While this GEMINI-TIA simulation can reproduce features seen in the observa-277 tions, a full system-level quantification of such a cusp event requires a more complete 278 data set to drive it. For this simulation we have chosen a localized frictional heating DCE 279 driver region consistent with the lower limit of imagery scale sizes shown in Burleigh et 280 al. (2019) and Lessard et al. (2019); the 0.3 deg extent of the simulated driver is also com-281 parable to the FWHM scale of the DCE event shown in Fig 2(a), assuming a spatial struc-282 ture. The simulated DCE has a strength $(40 \,\mathrm{mV/m})$ sufficient to raise the ion temper-283 ature through Joule heating by 300 K, the range of temperatures shown in Figure 3(a). 284 A full simulation of an event such as this requires multipoint in situ driving data at fine 285 resolution over a 2-D region, on the temporal and spatial scales and durations of the var-286 ious processes. Single-point in situ observations in the context of imagery can illustrate 287 features but it remains difficult to quantify the net effect of this intermittent driving, with 288 its intermittent responses, in terms of overall outflow, without reasonable inferences to 289 extend the fine-scale observations in space and time. Modern three-dimensional iono-290 spheric models such as GEMINI (Zettergren & Semeter, 2012) can model the ionospheric 291 heating volume. However, multipoint in situ observations that can separate the spatial 292 and temporal variations glimpsed by these observations and by those such as Oksavik 293 et al. (2004) and Moen et al. (2004), are needed to drive the model for conclusive quan-294 tification. A complete modelling quantification of the calculated upflow response requires 295 observations on a distributed grid, with observations covering the various spatial and tem-296 poral scales and durations of the heating process evolution illustrated in the simple sim-297 ulation here. A full characterization, thus, awaits the development of truly multipoint 298 in situ observations for both the drivers, and the responses. 299

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